

# Heat and Mass Transfer-2

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**2.1 CONDUCTION HEAT TRANSFER**

**2.2 CONVECTION HEAT TRANSFER**

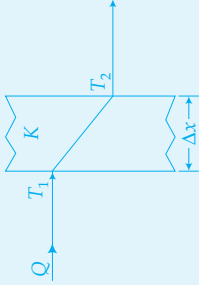
**2.3 RADIATION HEAT TRANSFER**

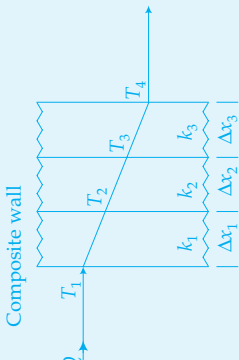
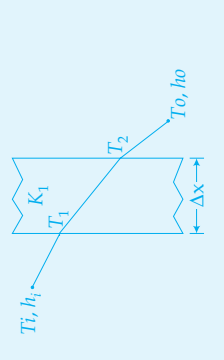
**2.4 MASS TRANSFER**

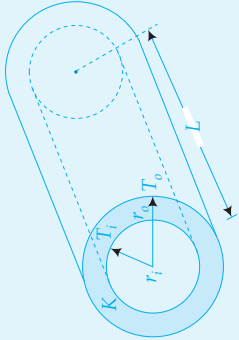
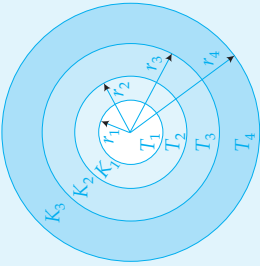
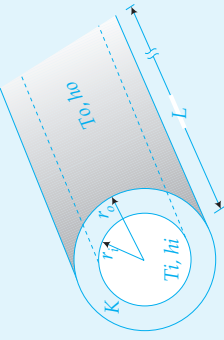
### 2.1.1.1 Steady State Conduction - One Dimension

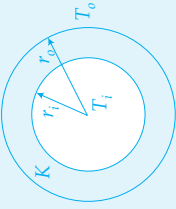
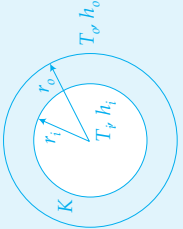
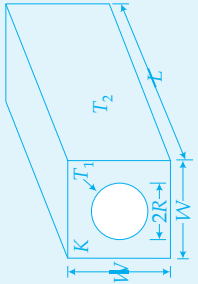
Sl. No.	Description	Correlation	Eq. No.
1.	General equation - one dimension Cartesian coordinates without heat generation With heat generation	$\frac{\partial^2 T}{\partial x^2} = 0$ $\frac{\partial^2 T}{\partial x^2} + \frac{\dot{q}}{K} = 0$	2.4a 2.4b
2.	Cylindrical coordinates no heat generation with heat generation	$\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} = 0$ $\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{\dot{q}}{K} = 0$	2.5a 2.5b

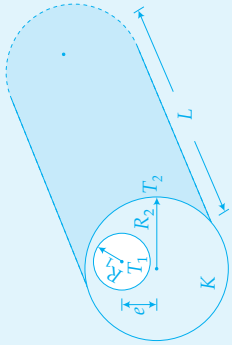
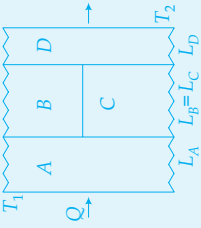
### 2.1.2 Conduction without Heat Generation

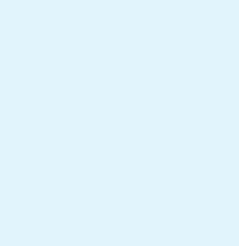
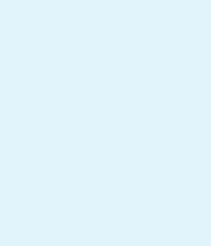
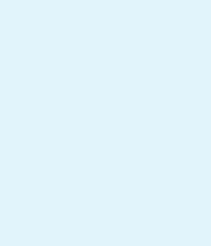
Sl. No.	Description	Correlation	Eq. No.	Notations
1.	Heat flow	$Q = \frac{\Delta T}{R}$	2.6	$\Delta T$ = Overall temperature difference, °C (or) K
2.	Plane wall 	$R = \frac{\Delta x}{KA}$	2.7	$R$ = Overall thermal resistance, K/W (or) °C/W $Q$ = Heat flow (or) Heat transfer, W $A$ = Area of heat flow, m <sup>2</sup>

3.	<p>Composite wall</p> 	$R = \frac{\Delta x_1}{k_1 A} + \frac{\Delta x_2}{k_2 A} + \frac{\Delta x_3}{k_3 A}$	<p>2.8</p> <p><math>h_i</math> = Inside heat transfer coefficient, W/m<sup>2</sup>K  <math>h_o</math> = Outside heat transfer coefficient, W/m<sup>2</sup>K  <math>U</math> = Overall heat transfer coefficient, W/m<sup>2</sup>K  <math>k_o</math> = Thermal conductivity at 0°C, W/mK  <math>\beta</math> = Co-efficient of thermal expansion</p>
4.	<p>Plane wall with convection ends</p> 	<p>2.9a</p> $R = \frac{1}{h_i A} + \frac{\Delta x}{k A} + \frac{1}{h_o A}$ <p>2.9b</p> $Q = UA(\Delta T)$ <p>where, <math>U = \frac{1}{A \cdot R}</math></p>	
5.	<p>Plane wall with variable thermal conductivity</p>	<p>2.10</p> $R = \frac{\Delta x}{k_m A}$ <p>where, <math>k_m = K_0 \left[ 1 + \beta \frac{(T_1 + T_2)}{2} \right]</math></p> <p>Replace <math>K</math> by <math>K_m</math> for linear variation of thermal conductivity</p>	

<p>6. Radial systems – cylinder</p> 	$R = \frac{\ln \frac{r_o}{r_i}}{2\pi KL}$ <p>2.11</p>	<p><math>r</math> = Radius, m <math>L</math> = Length of the cylinder, m</p>
<p>7. Multilayer cylinder</p>  <p>Length of the cylinder - <math>L</math></p>	$R = \frac{\ln \frac{r_2}{r_1}}{2\pi K_1 L} + \frac{\ln \frac{r_3}{r_2}}{2\pi K_2 L} + \frac{\ln \frac{r_4}{r_3}}{2\pi K_3 L}$ <p>2.12</p>	
<p>8. Cylinder with convective boundary conditions Heat flow</p> 	$R = \frac{1}{h_i A_i} + \frac{\ln \frac{r_o}{r_i}}{2\pi KL} + \frac{1}{h_o A_o}$ $Q = U_i A_i \Delta T = U_o A_o \Delta T$ <p>where, <math>U_i = \frac{1}{A_i R}</math> <math>U_o = \frac{1}{A_o R}</math></p> <p>2.13a 2.13b</p>	<p><math>A_i</math> = Inside area of the cylinder, <math>m^2 = 2\pi r_i L</math> <math>A_o</math> = Outside area of the cylinder, <math>m^2 = 2\pi r_o L</math></p>

9.	<p>Spherical systems - sphere</p> 	$R = \frac{1}{4\pi K} \left[ \frac{1}{r_i} - \frac{1}{r_o} \right]$	2.14
10.	<p>Sphere with convective boundary conditions</p> 	$R = \frac{1}{h_i A_i} + \frac{1}{4\pi K} \left[ \frac{1}{r_i} - \frac{1}{r_o} \right] + \frac{1}{h_o A_o}$ <p>where, <math>A_i = 4\pi r_i^2</math>  <math>A_o = 4\pi r_o^2</math></p>	<p><math>A_i</math> = Inside area of the sphere, m<sup>2</sup>  <math>A_o</math> = Outside area of the sphere, m<sup>2</sup></p>
11.	<p>Circular hole centered in a square solid of length L</p> 	$R = \frac{1}{2\pi K L} \ln \frac{0.54 W}{R} \quad \text{for } L \gg W$	2.16

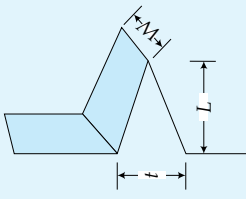
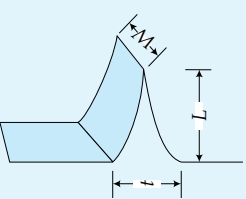
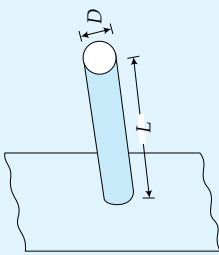
<p>12.</p>	<p>Eccentric circular hole (Radius, <math>R_1</math>) in a cylindrical solid (Radius, <math>R_2</math>) of length <math>L</math></p> 	$R = \frac{1}{2\pi KL} \cosh^{-1} [(R_1^2 + R_2^2 + e^2)/(eR_1R_2)]$ <p>2.17</p>	<p><math>K_A, K_B, K_C, K_D</math> = Thermal conductivity of A, B, C, D</p> <p><math>L</math> = Thickness</p> <p><math>A_A, A_B, A_C, A_D</math> = Area perpendicular to heat flow of A, B, C, D</p> <p><math>A_A = A_B + A_C = A_D</math></p>
<p>13.</p>	<p>Series - Parallel composite wall</p> 	<p>2.18</p> $R = R_1 + R_2 + R_3$ $R_1 = \frac{L_A}{K_A \cdot A_A}$ $R_2 = \frac{R_B R_C}{R_B + R_C}$ $R_B = \frac{L_B}{K_B \cdot A_B}, R_C = \frac{L_C}{K_C \cdot A_C}$ $R_3 = \frac{L_D}{K_D \cdot A_D}$	

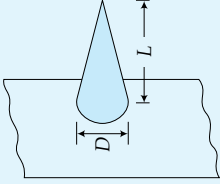
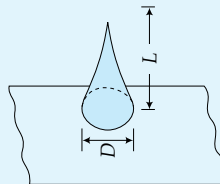
<p>3.</p> <p>Sphere with heat sources</p> 	$\frac{T_r - T_w}{T_o - T_w} = 1 - \left(\frac{r}{R}\right)^2$ $Q = \dot{q} \frac{4}{3} \pi R^3 = hA [T_w - T_\infty]$ $T_w = T_\infty + \frac{\dot{q}R}{3h}$ $T_o = T_w + \frac{\dot{q}R^2}{6K}$	<p>2.21a</p> <p>2.21b</p> <p>2.21c</p> <p>2.21d</p>
<p>4.</p> <p>Plane wall with asymmetric boundary condition</p> 	$T_x = \frac{\dot{q}L^2}{2K} \left[ 1 - \frac{x^2}{L^2} \right] + \frac{(T_{w2} - T_{w1})x}{2} \cdot \frac{x}{L}$ $+ \frac{(T_{w1} + T_{w2})}{2}$	<p>2.22</p>
<p>5.</p> <p>Hollow cylinder with asymmetric boundary condition</p> 	$T_r = T_o + \frac{\dot{q}}{4K} [R_o^2 - r^2] + \left[ (T_i - T_o) + \frac{\dot{q}[R_i^2 - R_o^2]}{4K} \ln \frac{r}{R_o} \right]$	<p>2.23</p>
<p>6.</p> <p>Hollow cylinder with inner surface adiabatic</p>	$T_r = T_o + \frac{\dot{q}}{4K} [R_o^2 - r^2] + \frac{\dot{q}}{2K} R_i^2 \ln \frac{r}{R_o}$	<p>2.24</p>
<p>7.</p> <p>Hollow cylinder with outside surface adiabatic</p>	$T_r = T_i + \frac{\dot{q}}{4K} [R_i^2 - r^2] + \frac{\dot{q}}{2K} R_o^2 \ln \left( \frac{r}{R_i} \right)$	<p>2.25</p>

2.1.4 Fins (or) Extended Surfaces

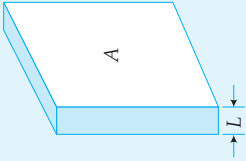
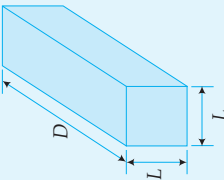
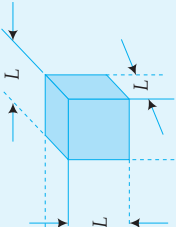
Sl. No.	Description	Correlation	Eq. No.	Notations
1.	<p>Rectangular fin</p>	$P = 2W + 2t$ $A = W.t$	<p>2.26a 2.26b</p>	<p><math>T_b</math> = Base temperature of the fin <math>T_\infty</math> = Temperature of the fluid <math>L</math> = Length of the fin, m <math>W</math> = Width of the fin, m <math>t</math> = Thickness of the fin, m <math>K</math> = Thermal conductivity of the fin, W/mK <math>h</math> = Convective heat transfer coefficient, W/m<sup>2</sup>K <math>\theta</math> = Excess temperature <math>= T - T_\infty</math> <math>\theta_b = T_b - T_\infty</math> <math>\theta_L = T_{x=L} - T_\infty</math> <math>\theta_0 = T - T_0</math></p>
2.	<p>Pin fin</p>	$P = \pi D$ $A = \frac{\pi}{4} D^2$ $m = \sqrt{\frac{hP}{KA}}$ $M = \sqrt{hPKA} \theta_0$	<p>2.27a 2.27b 2.27c 2.27d</p>	
3.	<p>Circumferential fins</p>	$\frac{\theta}{\theta_b} = \frac{I_0(mr) K_1(mr_2) + K_0(mr) I_1(mr_2)}{I_0(mr_1) K_1(mr_2) + K_0(mr_1) I_1(mr_2)}$ $Q = 2\pi K t \theta_b (mr_1) \left[ \frac{K_1(mr_1) I_1(mr_2) - I_1(mr_1) K_1(mr_2)}{I_0(mr_1) K_1(mr_2) + K_0(mr_1) I_1(mr_2)} \right]$ $\eta_f = \frac{2r_1/m}{(r_2^2 - r_1^2)} \left[ \frac{K_1(mr_1) I_1(mr_2) - I_1(mr_1) K_1(mr_2)}{I_0(mr_1) K_1(mr_2) + K_0(mr_1) I_1(mr_2)} \right]$ <p>where, <math>I_0</math> &amp; <math>K_0</math> are modified zero order Bessel functions of the first and second kind <math>I_1</math> &amp; <math>I_2</math> are modified first order Bessel functions of the first and second kind (refer Table 2.21)</p>	<p>2.28a 2.28b 2.28c</p>	
4.	<p>Circumferential, Rectangular, Triangular fins</p>	$Q = \eta_f A_f h [T_b - T_\infty]$	<p>2.29</p>	<p><math>\eta_f</math> = Fin efficiency <math>A_f</math> = Surface area of the fin, m<sup>2</sup></p>

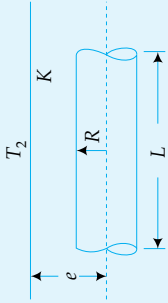
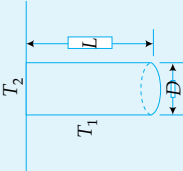
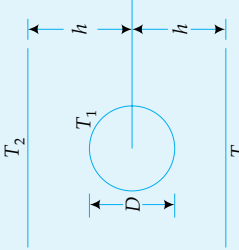


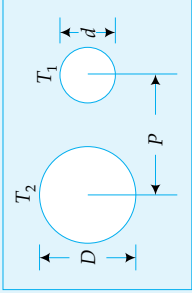
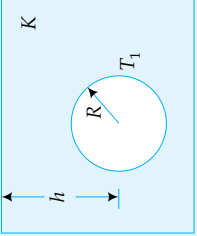
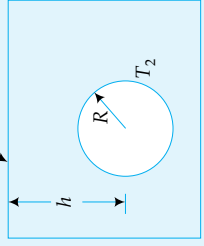
2.	<p>Triangular straight fin</p> 	$m = \sqrt{\frac{2h}{Kt}}$ $A_f = 2W \left[ L^2 + \left( \frac{t}{2} \right)^2 \right]^{1/2}$	$\frac{1}{mL} \frac{I_1(2mL)}{I_0(2mL)}$
3.	<p>Parabolic straight fin</p> 	$m = \sqrt{\frac{2h}{Kt}}$ $A_f = W \left[ C_1 L + \frac{L^2}{t} \ln \left( \frac{t}{L} + C_1 \right) \right]$ $C_1 = \left[ 1 + \left( \frac{t}{L} \right)^2 \right]^{1/2}$	$\frac{2}{\left[ 4(mL)^2 + 1 \right]^{1/2} + 1}$
4.	<p>Pin fin</p> 	$m = \sqrt{\frac{4h}{KD}}$ $A_f = \pi D L_C$ $L_C = L + \frac{D}{4}$	$\frac{\tanh mL_C}{m L_C}$

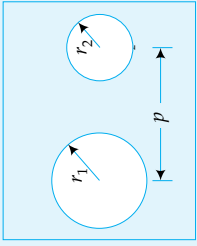
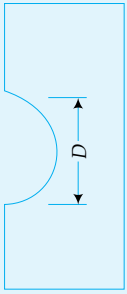
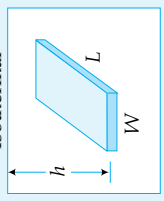
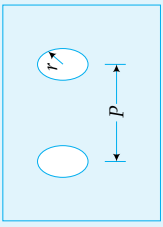
5.	<p>Triangular pin fin</p> 	$m = \sqrt{\frac{4h}{KD} \left[ L^2 + \left( \frac{D}{2} \right)^2 \right]^{\frac{1}{2}}}$ $A_f = \frac{\pi D}{2} \left[ L^2 + \left( \frac{D}{2} \right)^2 \right]^{\frac{1}{2}}$	$\frac{2 I_2(2 mL)}{mL I_1(2 mL)}$
6.	<p>Parabolic pin fin</p> 	$m = \sqrt{\frac{4h}{KD} \left[ C_3 C_4 - \frac{L}{2D} \ln \left( \frac{2DC_4}{L} + C_3 \right) \right]}$ $A_f = \frac{\pi L^3}{8D} \left[ C_3 C_4 - \frac{L}{2D} \ln \left( \frac{2DC_4}{L} + C_3 \right) \right]$ $C_3 = 1 + 2 \left( \frac{D}{L} \right)^2$ $C_4 = \left[ 1 + \left( \frac{D}{L} \right)^2 \right]^{\frac{1}{2}}$	$\frac{2}{\left[ \frac{4}{9} (mL)^2 + 1 \right]^{\frac{1}{2}} + 1}$

### 2.1.5 Multidirectional Steady State Conduction

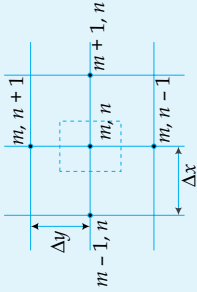
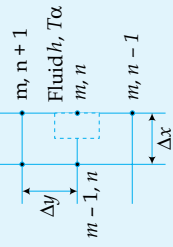
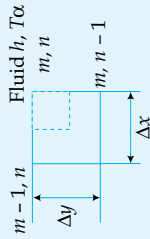
Sl. No.	Description	Correlation	Eq. No.	Notations
1.	Heat flow Conduction through a plane wall 	$Q = S \cdot K \cdot \Delta T$ $S = \frac{A}{L}$	2.30	$Q$ = Heat transfer, W $K$ = Thermal conductivity of the material, W/mK $\Delta T$ = Overall temperature difference $S$ = Shape factor, m
2.	Conduction through the edge of adjoining walls 	$S = 0.54D$ $D > \frac{L}{5}$	2.31	
3.	Conduction through corner of three walls with a temperature difference $\Delta T$ across the walls 	$S = 0.15L$	2.32	

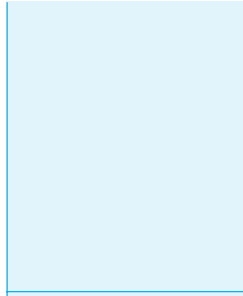
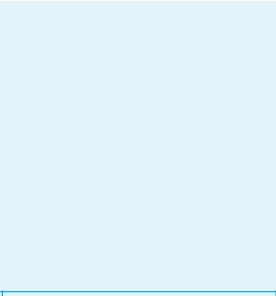
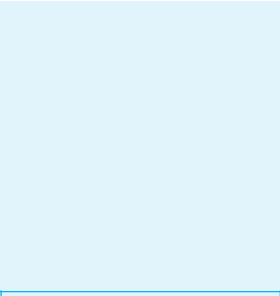
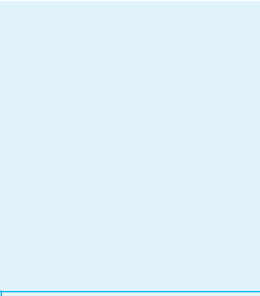
4.	<p>Isothermal cylinder of length <math>L</math>, at <math>T_1</math> placed horizontally in a semi-infinite medium having a surface temperature at <math>T_2</math></p> 	<p>2.33</p> $S = \frac{2\pi L}{\cosh^{-1}\left(\frac{h}{R}\right)}$ <p>for <math>L \gg R</math></p>	
5.	<p>Isothermal cylinder of length <math>L</math>, at <math>T_1</math> placed vertically in a semi-infinite medium having a surface at <math>T_2</math></p> 	<p>2.34</p> $S = \frac{2\pi L}{\ln\left(\frac{4L}{D}\right)}$ <p>for <math>L \gg D</math></p>	
6.	<p>A horizontal circular cylinder buried midway between two faces of a slab</p> 	<p>2.35</p> $S = \frac{2\pi L}{\ln\left(\frac{8h}{\pi D}\right)}$ <p>for <math>L \gg D, h &gt; \frac{D}{2}</math></p>	<p><math>L</math> = Length of the cylinder</p>

7.	<p>Conduction between two cylinders of diameters <math>D</math> and <math>d</math>, of length <math>L</math></p> 	$S = \frac{2\pi L}{\cosh^{-1} \left[ \frac{4p^2 - D^2 - d^2}{2Dd} \right]}$ <p>for <math>L \gg D, d, p</math></p>	2.36
8.	<p>An isothermal sphere at <math>T_1</math> placed in a semi-infinite medium having a surface temperature, <math>T_2</math></p> 	$S = \frac{4\pi R}{1 - \left( \frac{R}{2h} \right)}$ <p>for <math>h &gt; \frac{D}{2}</math></p>	2.37
9.	<p>An isothermal sphere at <math>T_1</math> placed near the insulated boundary of a semi-infinite medium at <math>T_2</math></p> 	$S = \frac{4\pi R}{1 + \left( \frac{R}{2h} \right)}$ <p>for <math>h &gt; \frac{D}{2}</math></p>	2.38

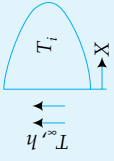
10.	<p>Two isothermal spheres buried in infinite medium</p> 	$S = \frac{4\pi}{\frac{r_2}{r_1} \left[ 1 - \frac{\left(\frac{r_1}{p}\right)^4}{1 - \left(\frac{r_2}{p}\right)^2} \right] - \frac{2r_2}{p}}$ <p>2.39</p>
11.	<p>Hemisphere buried in semi-infinite medium</p> 	$S = \pi D$ <p>2.40</p>
12.	<p>Thin rectangular plate of length L, buried in semi-infinite medium having isothermal surface</p> 	$S = \frac{2\pi W}{\ln\left(\frac{4W}{L}\right)}$ <p>for <math>h \gg W, W &gt; L</math></p> <p>2.41a</p> $S = \frac{\pi W}{\ln\left(\frac{4W}{L}\right)}$ <p>for <math>h = 0, W &gt; L</math></p> <p>2.41b</p> $S = \frac{2\pi W}{\ln\left(\frac{2\pi h}{L}\right)}$ <p>for <math>W \gg L, h &gt; 2W</math></p> <p>2.41c</p>
13.	<p>Parallel disks buried in infinite medium</p> 	$S = \frac{4\pi}{2 \left[ \frac{\pi}{2} - \tan^{-1} \left( \frac{r}{p} \right) \right]}$ <p>2.42</p>

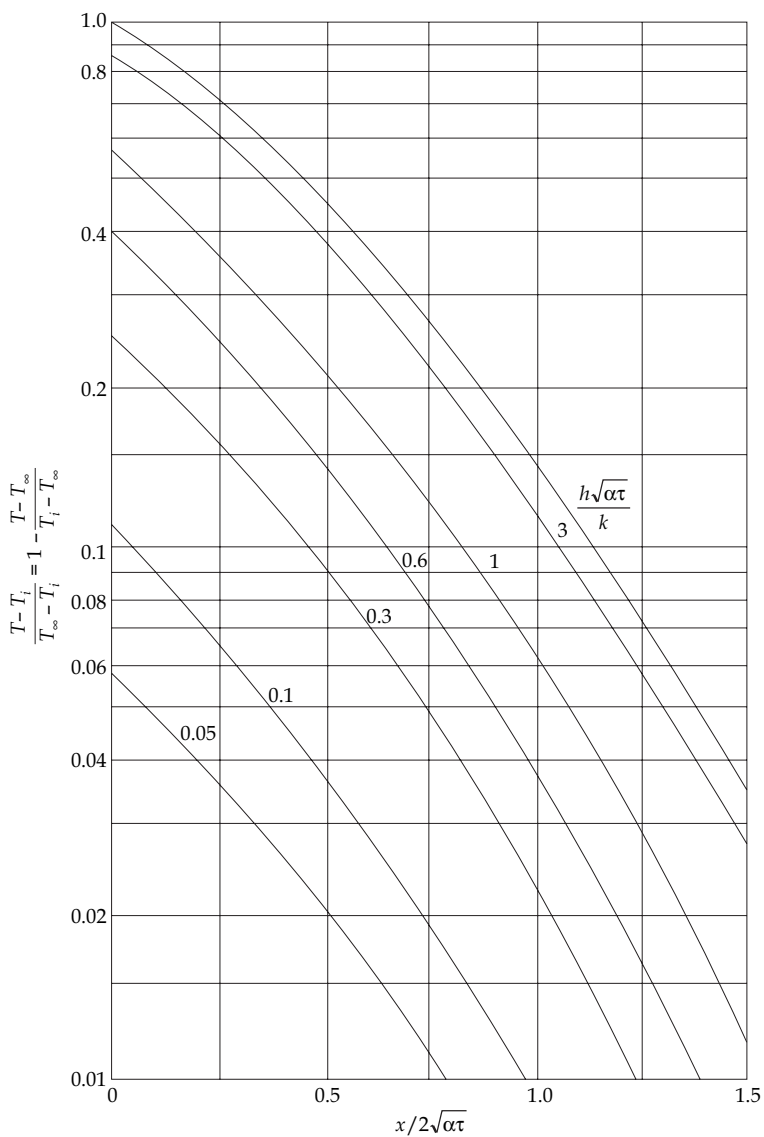
2.1.6 Numerical Analysis: Nodal Formulas for Finite difference Calculations ( $\Delta x = \Delta y$ )

Sl. No.	Description	Correlation	Eq. No.	Notations
1.	Steady state energy balance on an interior nodal point 	$T_{m-1,n} + T_{m+1,n} + T_{m,n-1} + T_{m,n+1} - 4T_{m,n} = 0$	2.46	$T$ = Temperature $m$ = Locations indicating the $x$ increment $n$ = Locations indicating the $y$ increment $h$ = Heat transfer coefficient, $W/m^2\text{ }^\circ\text{C}$ $K$ = Thermal conductivity, $W/m\text{ }^\circ\text{C}$ $T_\infty$ = Adjoining fluid temperature
2.	Surface node with convection boundary 	$2T_{m-1,n} + T_{m,n-1} + T_{m,n+1} + \frac{2h\Delta x}{K} T_\infty - 2\left[\frac{h\Delta x}{K} + 2\right] T_{m,n} = 0$	2.47	
3.	Exterior corner with convection boundary 	$T_{m,n-1} + T_{m-1,n} + 2\cdot\frac{h\Delta x}{K} T_\infty - 2\left[1 + \frac{h\Delta x}{K}\right] T_{m,n} = 0$	2.48	

<p>4. Interior corner with convection boundary</p> 	$2\left[T_{m-1,n} + T_{m,n+1}\right] + T_{m+1,n} + T_{m,n-1} + 2\frac{h_c \Delta x}{K} T_\infty - 2\left[3 + \frac{h_c \Delta x}{K}\right] T_{m,n} = 0$ <p style="text-align: right;">2.49</p>
<p>5. Node near curved boundary with specified temperature</p> 	$\frac{2}{1+a} T_{m+1,n} + \frac{2}{1+b} T_{m,n+1} + \frac{2T_1}{a(1+a)} + \frac{2T_2}{b(1+b)} - 2\left[\frac{1}{a} + \frac{1}{b}\right] T_{m,n} = 0$ <p style="text-align: right;">2.50</p>
<p>6. Node at a plane surface with insulated boundary</p> 	$T_{m,n+1} + T_{m,n-1} + 2T_{m-1,n} - 4T_{m,n} = 0$ <p style="text-align: right;">2.51</p>
<p>7. Node at a plane surface with uniform heat flux</p> 	$(2T_{m-1,n} + T_{m,n+1} + T_{m,n-1}) + \frac{2q_c \Delta x}{K} - 4T_{m,n} = 0$ <p style="text-align: right;">2.52</p>



5.	<p>Transient heat conduction with surface convection boundary conditions</p>  <p style="text-align: center;"><math>T_{(x,0)} = T_i</math></p> <p style="text-align: center;"><math>-K \frac{\partial T}{\partial x} \Big _{x=0} = h[T_{\infty} - T_{(0,t)}]</math></p>	$\frac{T_{(x,t)} - T_i}{T_{\infty} - T_i} = \text{erf} \left[ \frac{x}{2\sqrt{\alpha t}} \right] - \left[ \exp \left( \frac{hx}{K} + \frac{h^2 \alpha t}{K^2} \right) \right] \left[ \text{erf} \frac{x}{2\sqrt{\alpha t}} + \frac{h\sqrt{\alpha t}}{K} \right]$ <p style="text-align: right;">2.60</p>	
6.	Chart solutions: convection boundary conditions	$\frac{T_{(x,t)} - T_{\infty}}{T_i - T_{\infty}} = f \left[ \frac{x}{L}, \frac{hL}{K}, \frac{\alpha t}{L^2} \right]$ <p style="text-align: right;">2.61</p>	$T_0 =$ Temperature at mid plane $\frac{x}{L} = 0$
7.	Heisler charts	<p><math>\frac{T_0 - T_{\infty}}{T_i - T_{\infty}}</math> Vs <math>F_0</math> for various values of <math>1/B_i</math></p> <p><math>\frac{T_{(x,t)} - T_{\infty}}{T_0 - T_{\infty}}</math> Vs <math>\frac{1}{B_i}</math> for various values of <math>\frac{x}{L}</math></p> <p><math>\frac{T_{(x,t)} - T_{\infty}}{T_i - T_{\infty}} = \frac{T_0 - T_{\infty}}{T_i - T_{\infty}} \times \frac{T_{(x,t)} - T_{\infty}}{T_0 - T_{\infty}}</math> (refer Figure 2.4 to 2.9)</p> <p style="text-align: right;">2.62</p>	
8.	Grober charts	<p><math>\frac{Q}{Q_i}</math> Vs <math>\frac{h^2 \alpha t}{K^2}</math> for various values of <math>B_i</math> (refer Figure 2.10 to 2.12)</p> <p><math>Q_i = \rho c L [T_i - T_{\infty}]</math></p> <p style="text-align: right;">2.63</p>	$T_0 =$ Surface temperature i.e., at $x = 0$ at any time 't' $\omega = 2\pi f = \frac{2\pi}{t_0}$
9.	Systems with periodic variation	<p><math>\theta_0 = T_0 - T_m = \theta_{0,\max} \cos \omega t</math></p> <p><math>\theta_{(x,t)} = \theta_{0,\max} \exp \left[ -\sqrt{\omega/2\alpha} \cdot x \right] \cos \left[ \omega t - x\sqrt{\omega/2\alpha} \right]</math> 2.65</p> <p>where, <math>\theta_{0,\max} = T_{0,\max} - T_m</math></p> <p><math>Q = \frac{2}{\sqrt{\omega \alpha}} KA \theta_{0,\max}</math></p> <p style="text-align: right;">2.66</p>	$f =$ Frequency of temperature wave i.e., number of complete cycles per unit time $t_0 =$ Period of temperature oscillations $\alpha =$ Thermal diffusivity, $\text{m}^2/\text{sec}$ $K =$ Thermal conductivity, $\text{W/mk}$



**Fig. 2.3** Temperature distribution in the semi-infinite solid with convection boundary condition

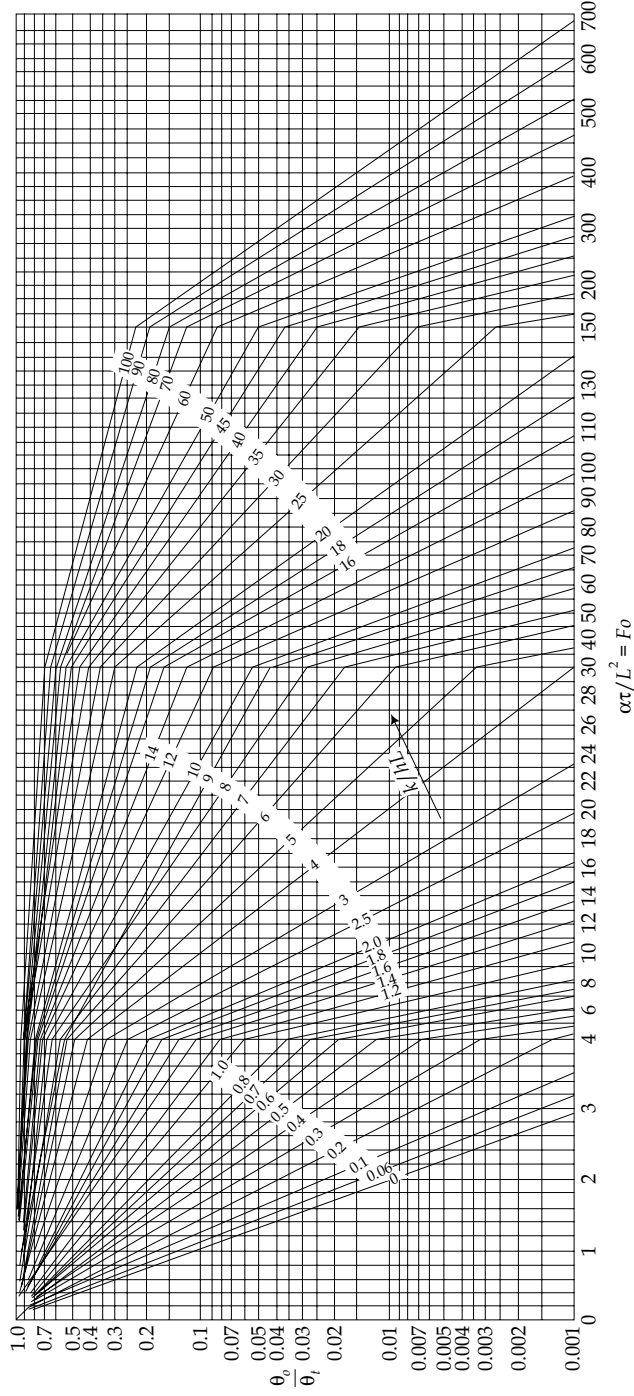


Fig. 2.4 Mid plane temperature for an infinite plate of thickness 2L

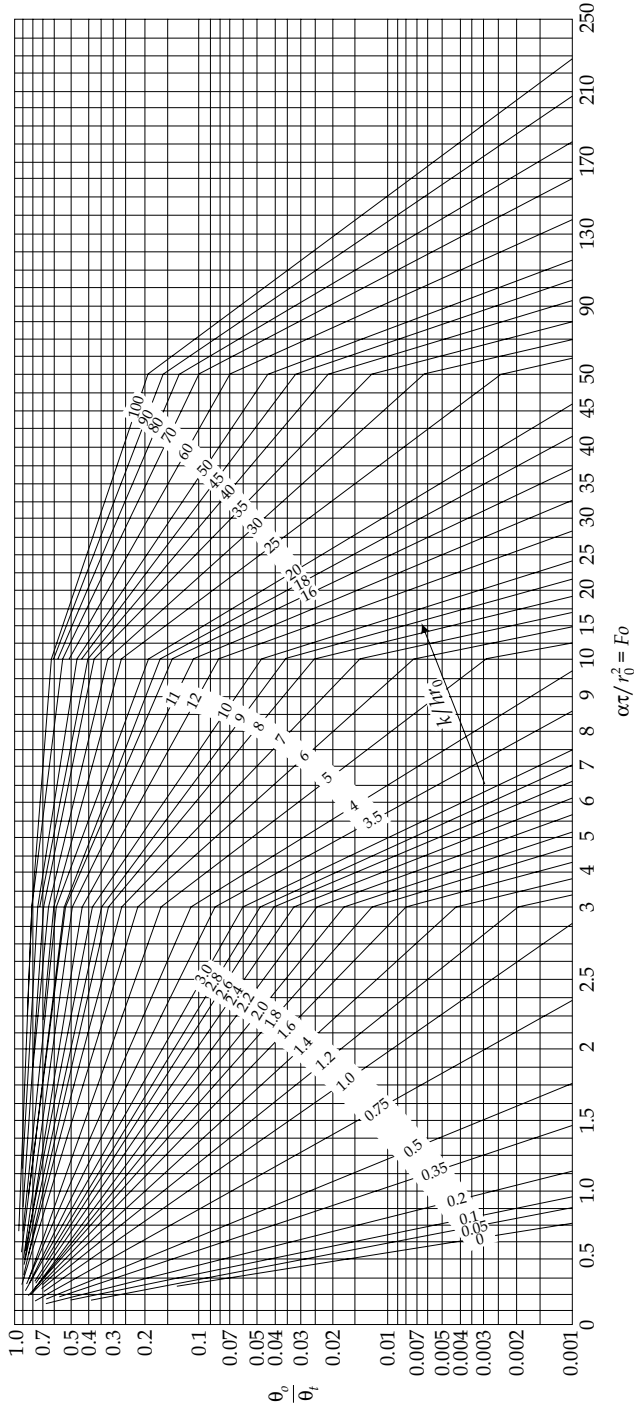
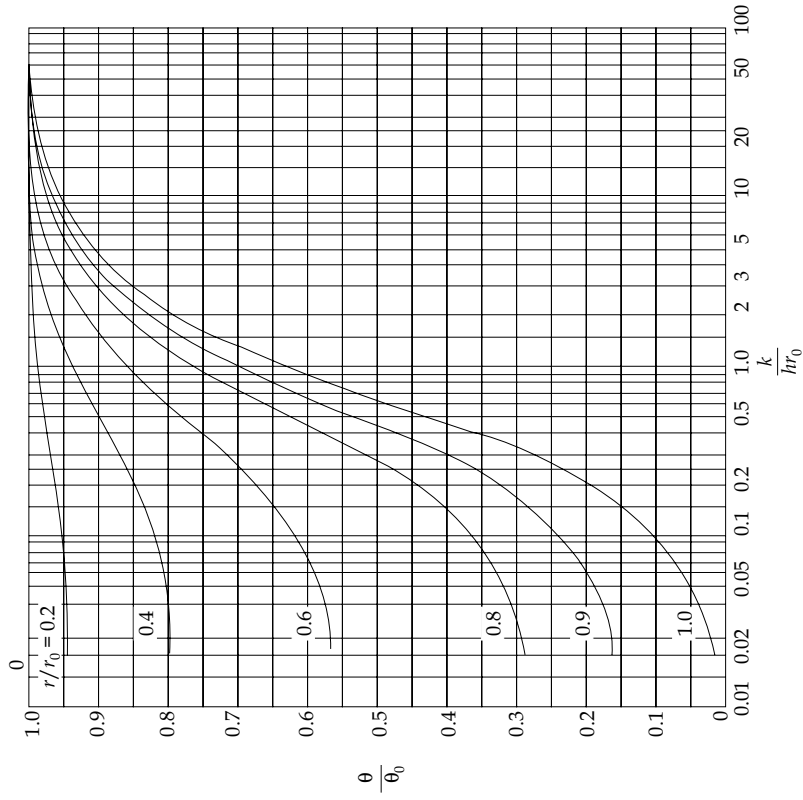
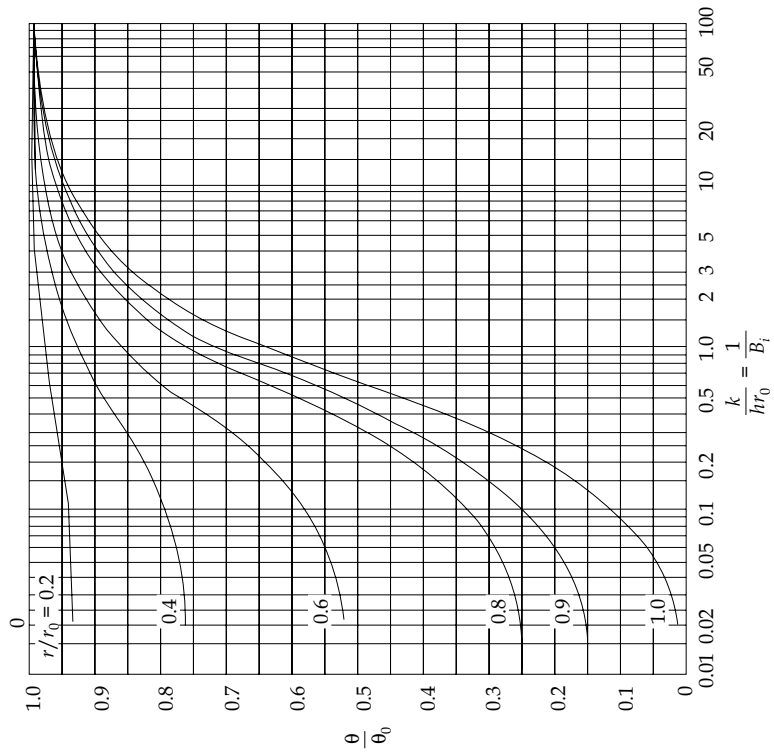


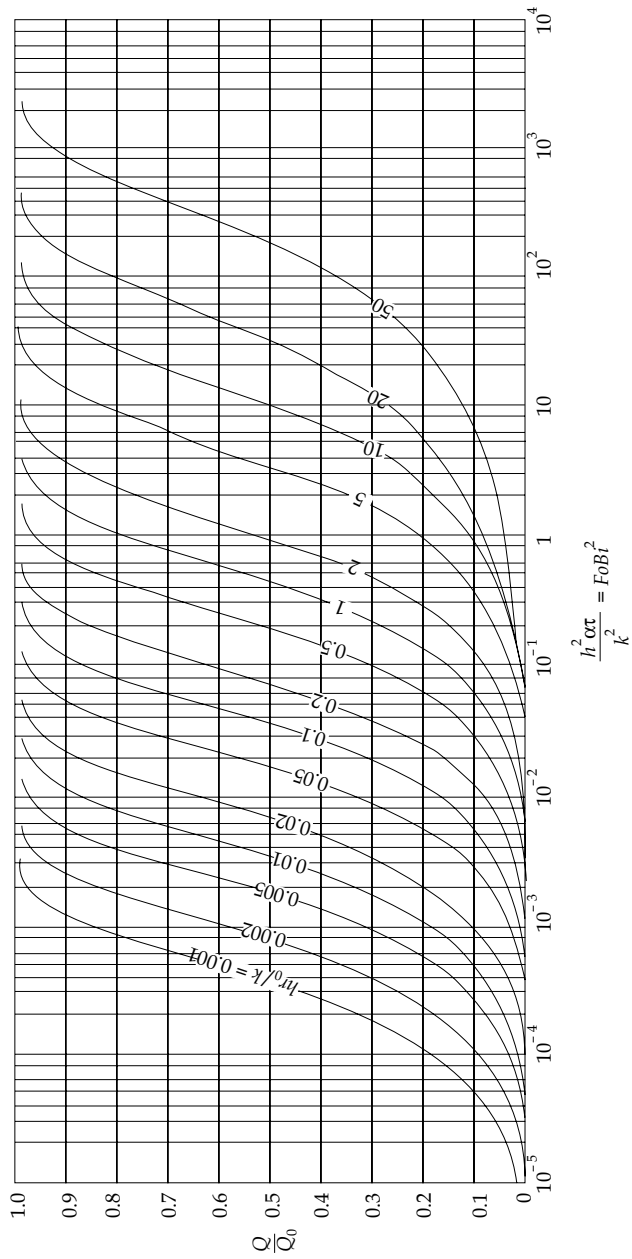
Fig. 2.6 Center temperature for a sphere of radius  $r_0$



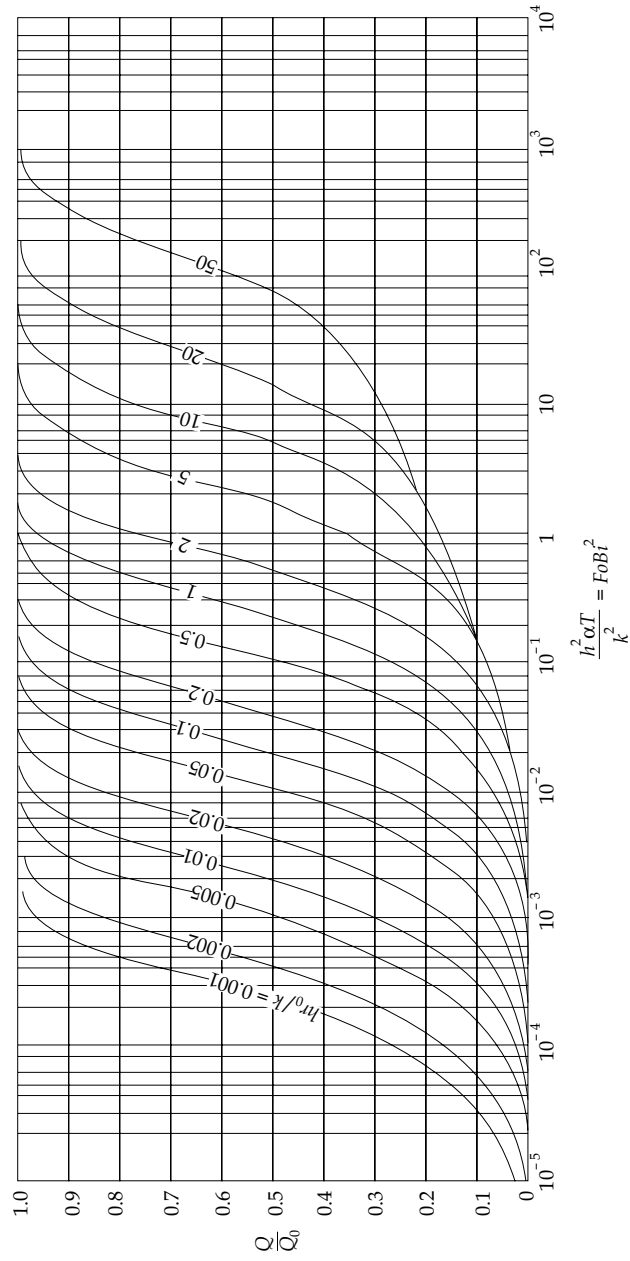
**Fig. 2.8** Temperature as a function of axis temperature in an infinite cylinder of radius  $r_0$



**Fig. 2.9** Temperature as a function of center temperature for a sphere of radius  $r_0$



**Fig. 2.11** Dimensionless heat loss  $Q/Q_0$  for an infinite cylinder of radius  $r_0$  with time



**Fig. 2.12** Dimensionless heat loss  $Q/Q_0$  for a sphere of radius  $r_0$  with time



## 2.2 CONVECTION HEAT TRANSFER



### 2.2.1 Dimensionless Groups










Sl. No.	Group Name	Group	Interpretation	Notations
1.	Coefficient of friction ( $C_f$ )	$\frac{\tau_s}{\rho V^2}$	Dimensionless shear stress	$\tau_s$ = Shear stress $\rho$ = Density, Kg/m <sup>3</sup> $V$ = Velocity, m/sec
2.	Friction factor ( $f$ )	$\frac{\Delta P}{\left(\frac{L}{d}\right)\left(\frac{\rho u_m^2}{2}\right)}$	Dimensionless pressure drop for internal flow	$\Delta P$ = Pressure drop $L$ = Length, m $d$ = Diameter, m $u_m$ = Mean flow velocity, m/sec
3.	Grashof number (Gr)	$\frac{g\beta(T_s - T_\infty)L^3}{\nu^2}$	Ratio of buoyancy to viscous forces	$g$ = Gravitational force, m/s <sup>2</sup> $\beta$ = coefficient of expansion, K <sup>-1</sup>
4.	Nusselt number (Nu)	$\frac{hL}{k}$	Dimensionless temperature gradient at the surface	$T_s$ = Surface temperature $T_\infty$ = Fluid temperature
5.	Peclet number (Pe)	$\frac{VL}{\alpha} = Re_L Pr$	Dimensionless Independent heat transfer parameter	$\nu$ = Kinematic viscosity, m <sup>2</sup> /sec $v = \mu/\rho$
6.	Prandtl number (Pr)	$\frac{\mu C_p}{k} = \frac{\nu}{\alpha}$	Ratio of the momentum and thermal diffusivities	$\mu$ = Dynamic viscosity, Kg/m sec
7.	Rayleigh number (Ra)	Gr.Pr		$h$ = Convective heat transfer coefficient, W/m <sup>2</sup> °C
8.	Reynolds number (Re)	$\frac{\rho VL}{\mu} = \frac{VL}{\nu}$	Ratio of Inertia and viscous forces	$K$ = Thermal conductivity, W/m°C $\alpha$ = Thermal diffusivity, m <sup>2</sup> /sec
9.	Stanton number (St)	$\frac{h}{\rho V C_p} = \frac{Nu_L}{Re_L \cdot Pr}$	Ratio of heat transfer coefficient to the heat flow per unit temperature rise	$C_p$ = Specific heat, kJ/kgK
10.	Graetz number (Gz)	$Re_L \cdot Pr \cdot \left(\frac{d}{L}\right)$		

2.2.2 Forced Convection - External Flow

External Flow: Properties are to be evaluated at  $T_f = [T_w + T_\infty]/2$  (unless otherwise stated)

Sl. No.	Flow Condition	Restrictions	Correlation	Eq. No.	Notations
<b>1. Flow Over Flat Plate:</b>					
a.	Laminar flow	$Re_x < 5 \times 10^5$			$x$ = Distance from leading edge
b.	Boundary layer thickness		$\delta = 5x Re_x^{-1/2}$	2.71	$Re_x$ = Reynolds number at a distance 'x'
c.	Local friction coefficient		$C_{fx} = 0.664 Re_x^{-1/2}$	2.72	$T_f$ = Film temperature
d.	Average friction coefficient		$\bar{C}_{fx} = 1.328 Re_x^{-1/2}$	2.73	$T_w$ = Plate surface temperature
e.	Thermal boundary layer thickness		$\delta_t = \delta Pr^{-1/3}$	2.74	$T_\infty$ = Free stream fluid temperature
f.	Turbulent flow	$5 \times 10^5 < Re_x < 10^9$			$Re_L$ = Reynolds number at a distance 'L'
g.	Boundary layer thickness		$\delta = 0.381 Re_x^{-1/5}$	2.75	$L$ = Length of the plate
h.	Local friction coefficient	$5 \times 10^5 < Re_x < 10^7$ $10^7 < Re_x < 10^9$	$C_{fx} = 0.0592 Re_x^{-1/5}$ $C_{fx} = 0.37 (\log Re_x)^{-2.594}$	2.76a 2.76b	$C_{fx}$ = Friction coefficient at a distance 'x'
i.	Average friction coefficient	$Re_L < 10^8$	$\bar{C}_{fL} = 0.074 Re_L^{-1/5} - 1742 Re_L^{-1}$	2.77	$\bar{C}_{fx}$ = Average friction coefficient at a distance 'x'
j.	Thermal boundary layer thickness		$\delta_t \sim \delta$	2.78	
<b>2. Flow over flat plate for constant wall temperature, Laminar flow (<math>T_w</math> = constant)</b>					
		$Re_x < 5 \times 10^5$	$Nu_x = 0.332 Re_x^{1/2} Pr^{1/3}$	2.79	$\bar{C}_{fL}$ = Average friction coefficient at a distance 'L'
		$0.5 \leq Pr \leq 50$	$\bar{Nu}_x = 0.664 Re_x^{1/2} Pr^{1/3}$	2.80	$Nu_x$ = Local Nusselt number at a distance of 'x'
		$Re_x < 5 \times 10^5$ $Re_x Pr > 100$	$Nu_x = \left[ \frac{0.3387 Re_x^{1/2} Pr^{1/3}}{1 + \left( \frac{0.0468}{Pr} \right)^{2/3}} \right]^{1/4}$	2.81	$\bar{Nu}_x$ = Average Nusselt number at a distance of 'x'
<b>3. For liquid metals</b>					
		$Re_x < 5 \times 10^5$ $Pr \ll 1$	$Nu_x = 0.564 (Re_x Pr)^{1/3}$	2.82	$St_x$ = Stanton number at a distance 'x'
<b>4. For constant surface heat flux (<math>q_w</math> = constant)</b>					
		$Re_x < 5 \times 10^5$ $0.6 < Pr < 50$	$Nu_x = 0.453 Re_x^{1/2} Pr^{1/3}$	2.83	Subscripts = x, L, D, ... . indicates the characteristic length
		$Re_x < 5 \times 10^5$	$Nu_x = \left[ \frac{0.4637 Re_x^{1/2} Pr^{1/3}}{1 + \left( \frac{0.0207}{Pr} \right)^{2/3}} \right]^{1/4}$	2.84	

			$10^3 - 2 \times 10^5$ $2 \times 10^5 - 10^6$ If $Pr \leq 10, n = 0.37$ $Pr > 10, n = 0.36$	0.26 0.076	0.6 0.7	
12. Cross flow of liquids across the cylinders		$0.1 < Re_D < 10^5$	$Nu_D = [0.35 + 0.56 Re_D^{0.52}] Pr^{0.3}$		2.97	
13. Cross flow of gases and liquids		$1 < Re_D < 10^3$	$Nu_D = (0.43 + 0.50 Re_D^{0.5}) Pr^{0.38} \left[ \frac{Pr_f}{Pr_w} \right]^{0.25}$		2.98	
		$10^3 < Re_D < 10^5$ $Pr_f$ at $T_f$ $Pr_w$ at $T_w$	$Nu_D = 0.25 Re_D^{0.6} Pr^{0.38} \left[ \frac{Pr_f}{Pr_w} \right]^{0.25}$		2.99	
14. Flow of air, water and liquid sodium		$10^2 < Re_D < 10^7$ $Pe_f > 0.2$	$Nu_D = 0.3 + \frac{0.62 Re_D^{1/2} Pr^{1/3}}{\left[ 1 + \left( \frac{0.4}{Pr} \right)^{2/3} \right]^{1/4}} \left[ 1 + \left( \frac{Re_D}{282,000} \right)^{5/8} \right]^{4/5}$		2.100	
		$40 < Re_D < 10^5$ $0.65 < Pr < 300$ $0.25 < \frac{\mu_\alpha}{\mu_w}$ $\mu_\alpha$ at $T_\infty$ $\mu_w$ at $T_w$	$Nu_D = [0.4 Re_D^{0.5} + 0.06 Re_D^{2/3}] \times Pr^{0.4} \left[ \frac{\mu_\alpha}{\mu_w} \right]^{0.25}$		2.101	
15. Flow over non-circular cylinders			$Nu_D = C Re_D^m Pr^{1/3}$		2.102	
	Geometry			Re <sub>D</sub>	C	m
	Square			$5 \times 10^3 - 10^5$	0.246	0.588
				$5 \times 10^3 - 10^5$	0.102	0.675

16. Flow of gases across a single, non-circular long cylinder of various geometries	<p>Hexagon</p> 	$5 \times 10^3 - 1.95 \times 10^4$ $1.95 \times 10^4 - 10^5$ $5 \times 10^3 - 10^5$	0.160 0.0385 0.153	0.638 0.782 0.638
	<p>Vertical Plate</p> 			
		$Nu_D = C Re_D^n$		2.103
<p>Geometry</p>		$Re_D$	$n$	$C$
		2500 - 7500 5000 - 100000	0.624 0.588	0.261 0.222
		2500 - 8000 5000 - 100000	0.699 0.675	0.160 0.092
		5000 - 100000	0.638	0.138
		5000 - 19500 19500 - 100000	0.638 0.782	0.144 0.035
		4000 - 15000	0.731	0.205
		3000 - 15000	0.804	0.085
		2500 - 15000	0.612	0.224

22. Flow across the tube bank		$N_L < 10$			$Nu_D = C_2 Nu_D (N_L \geq 10)$			2.112			$C_2 =$ Correction factor
$N_L$	1	2	3	4	5	6	7	8	9		
Aligned	0.64	0.80	0.87	0.90	0.92	0.94	0.96	0.98	0.99		
Staggered	0.68	0.75	0.83	0.89	0.92	0.95	0.97	0.98	0.99		
23. Flow across the tube bank		$Nu_L \geq 20$ $0.7 < Pr < 500$ $1000 < Re_D < 2 \times 10^6$ $Pr_w$ at $T_w$			$Nu_D = C Re_D^m Pr^{0.36} [Pr/Pr_w]^{1/4}$			2.113			
Configuration		$Re_D$			C			m			
Aligned		$10 - 10^2$			0.80			0.40			
Staggered		$10 - 10^2$			0.90			0.40			
Aligned ( $S_T/S_L > 0.7$ )		$10^3 - 2 \times 10^5$			0.27			0.63			
Staggered ( $S_T/S_L < 2$ )		$10^3 - 2 \times 10^5$			0.35 ( $S_T/S_L$ ) <sup>1/5</sup>			0.60			
Staggered ( $S_T/S_L > 2$ )		$10^3 - 2 \times 10^5$			0.40			0.60			
Aligned		$2 \times 10^5 - 2 \times 10^6$			0.021			0.84			
Staggered		$2 \times 10^5 - 2 \times 10^6$			0.022			0.84			
24. Flow through a packed bed and a fluidized bed		$90 \leq Re_i \leq 4000$			$St(Pr)^{2/3} = \frac{2.06}{\epsilon} (Re_D)^{-0.575}$			2.114			$d =$ Effective diameter of a particle
25. Long flate plate, width 'D' perpendicular to flow in a gas		$1 < Re_D < 4 \times 10^5$			$Nu_D = 0.20 Re_D^{2/3}$			2.115			$\epsilon =$ Bed porosity or void fraction = 0.3 to 0.5
26. Half-round cylinder with flat rear surface in a gas		$1 < Re_D < 4 \times 10^5$			$Nu_D = 0.16 Re_D^{2/3}$			2.116			
27. Square plate, dimension $L$ , perpendicular to flow of gas or a liquid		$2 \times 10^4 < Re_L < 10^5$			$St_L Pr^{2/3} = 0.930 Re_L^{-1/2}$			2.117			
28. Heat transfer in High speed flow-flow in aircraft and missiles		$Re_x < 10^5$ (Laminar)			$St_x = 0.332 (Re_x)^{-1/2} (Pr)^{-2/3}$			2.118			$T_{as} =$ Adiabatic surface temperature
		$10^5 < Re_x < 10^7$ (Turbulent)			$St_x = 0.0288 (Re_x)^{-1/5} (Pr)^{-2/3}$			2.119			
		$10^7 < Re_x < 10^9$ (Turbulent)			$St_x = [2.46 / \ln Re_x]^{2.58} (Pr)^{-2/3}$			2.120			Thermo Physical Properties are evaluated at $T = T_w + 0.5 (T_w - T_\infty) + 0.22 (T_{as} - T_\infty)$

2.2.3 Forced Convection - Internal Flow

Internal Flow: Properties are evaluated at bulk mean temperature  $T_b = (T_i + T_o)/2$

Sl. No.	Flow Condition	Restrictions	Correlation	Eq. No.	Notations
1.	Tube Flow: Laminar flow Friction factor	$Re_D < 2300$	$f = 64/Re_D$	2.121	$D$ = Diameter of the tube $L$ = Length of the tube $T_i$ = Inlet flow temperature $T_o$ = Outlet flow temperature $T_w$ = Surface temperature of the tube $T_b$ = Bulk mean temperature
2.	Laminar flow, fully developed with uniform heat flux ( $q = \text{constant}$ )		$Nu_D = 4.36$	2.122	
3.	Laminar flow, fully developed with uniform surface temperature ( $T_w = \text{constant}$ )		$Nu_D = 3.66$	2.123	
4.	Laminar flow in the entrance region of a circular tube with constant surface temperature		$Nu_D = 3.66 + \frac{0.0668 \left(\frac{D}{L}\right) Re_D Pr}{1 + 0.04 \left[\left(\frac{D}{L}\right) Re_D Pr\right]^2}$	2.124	
5.	Laminar flow in a circular tube at constant surface temperature	$0.48 < Pr < 16,700$ $0.004 < \left(\frac{\mu}{\mu_w}\right) < 9.75$ $\mu_s$ at $T_w$	$Nu_D = 1.86 \left[\frac{Re_D Pr}{L/D}\right]^{1/3} \left[\frac{\mu}{\mu_w}\right]^{0.14}$	2.125	
6.	Turbulent flow Fully developed turbulent flow	$Re_D > 2300$ $0.6 < Pr < 100$	$Nu_D = 0.023 Re_D^{0.8} Pr^n$ $n = 0.4$ for heating $n = 0.3$ for cooling	2.126	
		$0.7 \leq Pr \leq 16,700$ $Re_D \geq 10,000$ $\mu_w$ at $T_w$	$Nu_D = 0.027 Re_D^{4/5} Pr^{1/3} \left(\frac{\mu}{\mu_w}\right)^{0.14}$	2.127	
		$0.5 < Pr < 2000$ $3000 < Re_D < 5 \times 10^6$	$Nu_D = \frac{\left(\frac{f}{\theta}\right) (Re_D - 1000) Pr}{1 + 12.7 \left(\frac{f}{\theta}\right)^{1/2} (Pr^{2/3} - 1)}$	2.128	

14. Flow through the concentric tube annulus - fully developed turbulent flow		$Nu_{Dh} = 0.023 Re_{Dh}^{4/5} Pr^n$ $n = 0.4$ for heating $n = 0.3$ for cooling $D_h = D_o - D_i$	2.136	$A =$ Cross-sectional area of flow $P =$ Wetted perimeter
15. Flow through the non-circular tubes		$Nu_{Dh} = 0.023 Re_{Dh}^{4/5} Pr^4$ $n = 0.4$ for heating $n = 0.3$ for cooling $D_h = \frac{4A}{P}$	2.137	

### 2.2.4 Natural (or) Free Convection

Properties are evaluated at film temperature  $T_f = \frac{1}{2}[T_w + T_\infty]$  [Unless otherwise stated]

Sl. No.	Flow Condition	Restrictions	Correlation	Eq. No.	Notations	
1.	Free convection from vertical flat plates and cylinders with uniform wall temperature.		$Nu_L = C [Gr_L Pr]^m$	2.138	$L =$ Height of the plate or cylinder $T_w =$ Temperature of the wall surface of the wall or cylinder $T_\infty =$ Temperature of the fluid	
		$Gr_L Pr$	C			m
		$10^4 - 10^9$ (Laminar flow) $10^9 - 10^{13}$ (Turbulent flow) $10^9 - 10^{13}$	$0.59$ $0.021$ $0.10$			$\frac{1}{4}$ $\frac{2}{5}$ $\frac{1}{3}$
2.	Free convection from an isothermal vertical plate or cylinder	$10^{-1} < Ra_L < 10^{12}$	$Nu_L = \left\{ 0.825 + \frac{0.387 Ra_L^{1/6}}{\left[ 1 + \left( \frac{0.492}{Pr} \right)^{9/16} \right]^{8/27}} \right\}^2$	2.139		

3. Free convection on a vertical wall of height $L$ subjected to a uniform heat flux ' $q$ '	$Ra_L < 10^9$	$Nu_L = 0.68 + \frac{0.670 Ra_L^{1/4}}{\left[1 + \left(\frac{0.492}{Pr}\right)^{9/16}\right]^{4/9}}$	2.140	$K$ = Thermal conductivity $\nu$ = Kinematic viscosity $\beta$ = Coefficient of thermal expansion $g$ = Acceleration due to gravity = 9.81 m/sec <sup>2</sup>	
	$2 \times 10^3 < Ra_L < 10^{16}$	$Nu_L = 0.17 (Gr_L Pr)^{1/4}$	2.141		
4. Free convection on a vertical plate with uniform heat flux ' $q$ '	$10^5 < Gr_L^* Pr < 10^{11}$ (Laminar)	$Nu_L = 0.75 (Gr_L^* Pr)^{1/5}$	2.142		
	$2 \times 10^{13} < Gr_L^* Pr < 10^{16}$ (Turbulent)	$Nu_L = 0.645 (Gr_L^* Pr)^{0.22}$ Where, $Gr_L^*$ modified Grashof number $Gr_L^* = \frac{\beta q L^4}{Kg^2}$	2.143		
	$Pr < 0.03$	$Nu_L = 0.68 (Gr_L Pr^2)^{1/4}$	2.144		
5. For liquid metals	$10^4 < Gr_L Pr < 10^9$	$Nu_L = 0.52 (Gr_L Pr)^{1/4}$	2.145		$l$ = Characteristic length = distance fluid particle travels in boundary layer $A$ = Surface area of the plate $P$ = Perimeter
6. For Irregular solids	$10^4 < Gr_D Pr < 10^6$	$Nu_D = 0.775 (Gr_D Pr)^{0.21}$	2.146		
7. Free convection on a vertical cylinder	$10^4 \leq Gr_L Pr \leq 10^7$	$Nu_L = 0.54 (Gr_L Pr)^{1/4}$	2.147a		
	$10^7 \leq Gr_L Pr \leq 10^{11}$ $T_W = \text{Constant}$	$Nu_L = 0.15 (Gr_L Pr)^{1/3}$ Characteristic length $L = \frac{A}{P}$	2.147b		
9. Hot surface facing down (or) Cold surface facing up	$10^5 \leq Gr_L Pr \leq 10^{10}$ $T_W = \text{Constant}$	$Nu_L = 0.27 (Gr_L Pr)^{1/4}$ Characteristic length $L = \frac{A}{P}$	2.148		
	$Gr_L Pr < 2 \times 10^8$	$Nu_L = 0.13 (Gr_L Pr)^{1/3}$	2.149a		
10. Free convection on a horizontal plate subjected to uniform surface heat flux ' $q$ ' heated surface facing upward	$10^8 \leq Gr_L Pr < 10^{11}$	$Nu_L = 0.16 (Gr_L Pr)^{1/3}$	2.149b		
	$10^6 \leq Gr_L Pr < 10^{11}$	$Nu_L = 0.58 (Gr_L Pr)^{1/5}$ Properties are to be evaluated at $T_f = T_w - 0.25 (T_w - T_\infty)$ and $\beta$ at $(T_w + T_\infty)/2$	2.150		
11. Heated surface facing downward					



18. Free convection heat transfer from spheres to air	$1 < Gr_D, Pr < 10^5$	$Nu_D = 2 + 0.43(Gr_D Pr)^{1/4}$	2.158	$Ra = Gr.Pr$ $Ra =$ Rayleigh number
19. Spheres with water	$3 \times 10^5 < Gr_D, Pr < 8 \times 10^8$	$Nu_D = 2 + 0.5(Gr_D Pr)^{1/4}$	2.159	$Ra_W =$ Rayleigh number with characteristic length $W$
20. Free convection in rectangular cavities	$10^4 < Ra_W < 10^7$ $1 < Pr < 2 \times 10^4$	$Nu_W = 0.42(Ra_W)^{0.25} (Pr)^{0.012} \left(\frac{L}{W}\right)^{-0.3}$	2.160a	$L =$ Length of the rectangular cavity
	$10 < \frac{L}{W} < 40$			$W =$ Width of the enclosure
21. Free convection in concentric cylinders enclosure	$10^6 < Ra_W < 10^9$ $1 < Pr < 20$	$Nu_W = 0.046(Ra_W)^{1/3}$	2.160b	(Characteristic length)
	$1 < \frac{L}{W} < 40$			
21. Free convection in concentric cylinders enclosure	$10^2 < Ra_c < 10^7$	$q = \frac{2\pi k_{eff} [T_i - T_o]}{\ln\left(\frac{d_o}{d_i}\right)}$	2.161	$d_i =$ Outer diameter of the inner tube
		$\frac{k_{eff}}{k} = 0.386 \left[ \frac{Pr}{0.861 + Pr} \right]^{0.25} [Ra_c^*]$		$d_o =$ Inner diameter of the outer tube
		$Ra_c = \frac{\left[ \ln\left(\frac{d_o}{d_i}\right) \right]^{-4}}{L^3 \left[ d_i^{-0.6} + d_o^{-0.6} \right]^5} Ra_L$		$L = d_o - d_i =$ The characteristic length of the annular enclosure
		$Ra_L = \frac{g\beta [T_i - T_o] L^3}{\gamma^2} . Pr$		$k_{eff} =$ Effective thermal conductivity
		$L = \frac{1}{2} (d_o - d_i)$		$q =$ Heat flux (or) rate of heat flow
				$T_i$ & $T_o$ are the temperatures of the inner and the outer cylinder walls

<p>22. Free convection heat transfer between concentric spheres</p>		$q = K_{eff} \pi \left[ \frac{d_i d_o}{L} \right] [T_i - T_o]$ $\frac{K_{eff}}{K} = 0.74 \left[ \frac{\text{Pr}}{0.861 + \text{Pr}} \right]^{1/4} [Ra_C^*]^{1/4}$ $Ra_C^* = \left[ \frac{L}{(d_o d_i)^4} \frac{Ra_L}{[d_i^{-7/5} + d_o^{-7/5}]^5} \right]$ $Ra_L = \frac{\delta \beta [T_i - T_o] L^3}{\gamma^2} \cdot \text{Pr}$ $L = \frac{1}{2} (d_o - d_i)$	<p>2.162</p>	
<p>23. Combined free and forced convection</p>		$Nu^3 = [Nu_{forced}]^3 \pm [Nu_{free}]^3$	<p>2.163</p>	
<p>24. Natural convection between long rotating cylinder and a surrounding fluid</p>	<p><math>Re_w &gt; 800</math></p>	$Nu_D = 0.11 (0.5 Re_w^2 + Gr_D \cdot \text{Pr})^{0.35}$	<p>2.164</p>	<p><math>\omega</math> = Angular velocity, (or) rotating speed, rad/sec</p>
<p>25. Natural convection between rotating disk and a surrounding fluid</p>	<p><math>Re_w &lt; 5 \times 10^5</math></p>	$Nu_D = 0.35 (Re_w)^{1/2}$	<p>2.165</p>	$Re_w = \frac{\omega \pi D^2}{\gamma}$
<p>26. Natural convection between rotating sphere and a surrounding fluid</p>	<p><math>Re_w &lt; 5 \times 10^4</math> <math>\text{Pr} &gt; 0.7</math> <math>5 \times 10^4 &lt; Re_w &lt; 7 \times 10^5</math></p>	$Nu_D = 0.43 Re_w^{0.5} \text{Pr}^{0.4}$	<p>2.166a</p>	<p><math>D</math> = Diameter of the cylinder/disc/sphere</p>
		$Nu_D = 0.066 Re_w^{0.67} \text{Pr}^{0.4}$	<p>2.166b</p>	

2.	Critical heat flux for nucleate pool boiling	$q_{\max} = 0.149 h_{fg} \rho_w \left[ \frac{\sigma_s g (\rho_l - \rho_w)}{\rho_w^2} \right]^{1/4}$	2.168	$k_l$ = Thermal conductivity of saturated liquid, W/m°C $(\Delta T)_e$ = Excess Temperature $= T_w - T_{Sat}$ $T_w, T_{Sat}$ = Temperature of wall and Saturation Temperature, respectively $k_v$ = Thermal conductivity for Saturated vapour, W/m°C $C_{pw}$ = Specific heat of Saturated vapour, J/Kg°C $\mu_w$ = Vapour viscosity, Kg/m.s $d$ = Tube diameter $h_b$ = Average heat transfer coefficient in the absence of radiation, W/m <sup>2</sup> °C $h$ = Total heat transfer coefficient, W/m <sup>2</sup> °C $h_r$ = Radiation heat transfer coefficient, W/m <sup>2</sup> °C $\epsilon$ = Emissivity of the surface $\sigma_s$ = Stefan Boltzmann constant = $5.67 \times 10^{-8}$ W/m <sup>2</sup> K <sup>4</sup> $q$ = Heat flux, KW/m <sup>2</sup> $h_p$ = Heat transfer coefficient at some pressure 'p' $P$ = System Pressure $P_1$ = Standard atmospheric pressure
3.	Minimum heat flux from a large horizontal plate	$q_{\min} = C_p \rho_w h_{fg} \left[ \frac{g \sigma_s (\rho_l - \rho_w)}{(\rho_l + \rho_w)^2} \right]^{1/4}$ <p>where, <math>c</math> is constant equals to 0.09</p> $h_b = 0.62 \left[ \frac{K_v^3 \rho_w (\rho_l - \rho_w) g (h_{fg} + 0.4 C_{pw} \Delta T_e)}{\mu_w d \Delta T_e} \right]^{1/4}$	2.169	
4.	Stable film boiling (or) film pool boiling in the absence of radiation	$h = h_b \left[ \frac{h_b}{h} \right]^{-0.33} + h_r$	2.171	
5.	Considering radiative heat transfer	$h_r = \frac{\sigma_s \epsilon [T_w^4 - T_{sat}^4]}{(T_w - T_{sat})}$	2.172	
6.	Simplified relations for boiling heat transfer with water at atmospheric pressure	Surface	$q, \text{ KW/m}^2$ $h, \text{ W/m}^2\text{°C}$ 2.173	
		Horizontal	$q < 16$ $1042 (\Delta T_e)^{1/3}$ $16 < q < 240$ $5.56 (\Delta T_e)^3$	
		Vertical	$q < 3$ $537 (\Delta T_e)^{1/7}$ $3 < q < 63$ $7.96 (\Delta T_e)^3$	
7.	Boiling heat transfer with water at different pressures	$h_p = h \left[ \frac{P}{P_1} \right]^{-0.4}$	2.174	

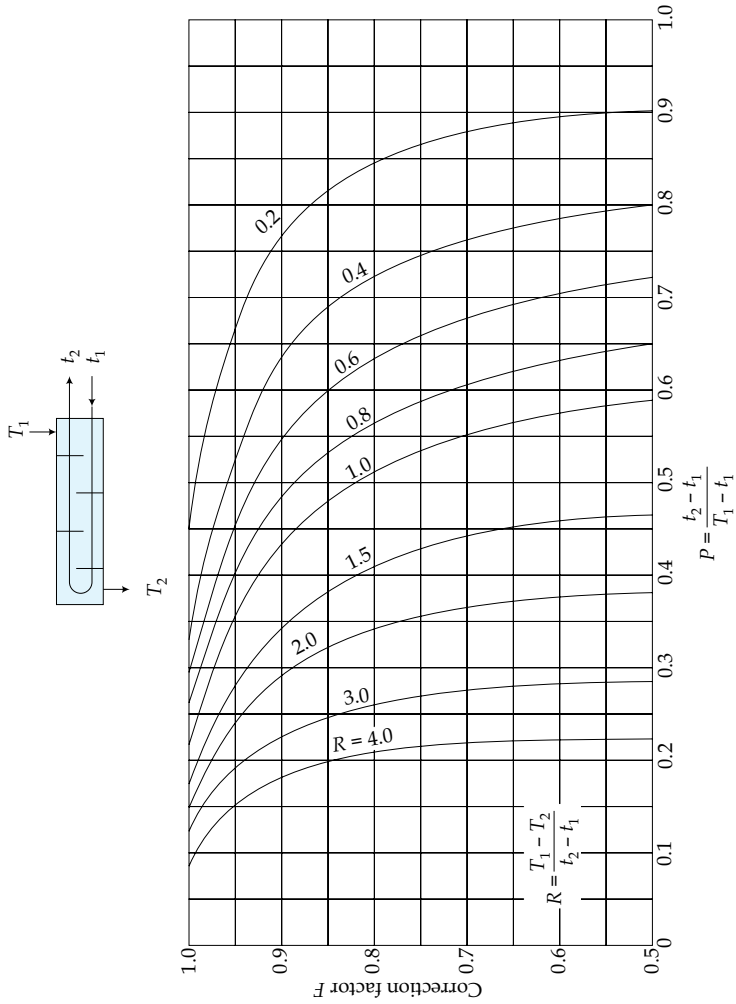
## 2.2.6 Condensation

Sl. No.	Description	Correlation	Eq. No.	Notations
1.	Laminar film wise condensation on a vertical plate	$h_{mf} = 1.13 \left[ \frac{g \rho_l (\rho_l - \rho_w) h_{fg} K_f^3}{\mu_l [T_v - T_w] L} \right]^{1/4}$ <p>for <math>Re &lt; 1800</math>            Physical Properties are evaluated at the film temperature  <math>T_f = \frac{1}{2} [T_w + T_v]</math></p>	2.175	$h_{fg}$ = Latent heat of condensation J/Kg $L$ = Length of vertical plate $T_v, T_w$ = Vapour and wall temperature, respectively $\dot{m}$ = Mass flow rate of condensate, Kg/sec $P$ = Wetted Perimeter $A_t$ = Condensing surface area, $m^2$ $h_{m,av}$ = Average condensation heat transfer coefficient $W/m^2 \cdot ^\circ C$ $N$ = Number of tubes $L$ = Length of the tube $h_{m,vert} = h_{m,horz}$ for vertical tube $h_{m,horz} = h_{m,vert}$ for horizontal tube
2.	Film condensation on a vertical plate for turbulent flow	$h_{mt} = 0.0077 (Re)^{0.4} \left[ \frac{K_f^3 \rho_l^2 g}{\mu_l^2} \right]^{1/3}$ <p>for <math>Re &gt; 1800</math>            Reynolds number <math>Re = \frac{4 \dot{m}}{\mu_l P}</math>  <math>P</math> = Wetted perimeter for a vertical plate of width '<math>w</math>'  <math>= \pi D</math> for a vertical tube of outside diameter <math>D</math>  <math>= 2L</math> for horizontal tubes each of length <math>L</math> arranged in vertical tiers  <math display="block">\dot{m} = \frac{A_t h_{m,av} [T_v - T_w]}{h_{fg}}</math></p>	2.176	
3.	Film condensation on Inclined plates	replace $g$ by $g \cos \theta$ in above equations		
4.	Film condensation on horizontal tube and on horizontal tube banks	$h_{mt} = 0.725 \left[ \frac{g \rho_l (\rho_l - \rho_w) h_{fg} K_f^3}{\mu_l [T_v - T_w] N d} \right]^{1/4}$	2.177	
5.	Comparison for film wise condensation on a vertical tube of length ' $L$ ' and a horizontal tube of diameter ' $d$ '	$\frac{h_{m,vert}}{h_{m,horz}} = 1.3 \left[ \frac{d}{L} \right]^{1/4}$	2.178	

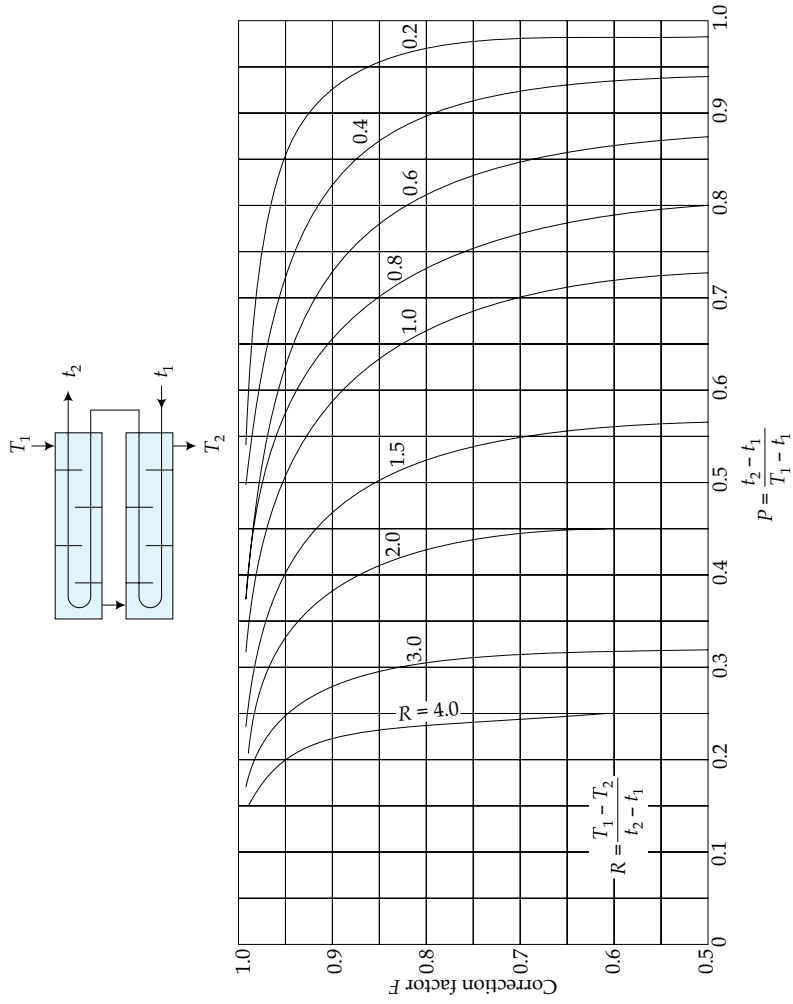
10.	If $C_h < C_c$	$\epsilon = \frac{T_{hi} - T_{ho}}{T_{hi} - T_{ci}}$	2.189	$C_{hr}, C_c$ = Heat capacity of hot and cold fluid, W/k $N$ = Number of Transfer Units (NTU) $f$ = Friction factor $V_t$ = Mass velocity of the tube fluid, Kg/m <sup>2</sup> s $L$ = Length of the tube $n$ = Number of tube passes $g$ = Gravitational acceleration, 9.8 m/s <sup>2</sup> $\rho_t$ = Density of tube fluid, Kg/m <sup>3</sup> $d_i$ = Inside diameter of a tube $\phi$ = Viscosity correction factor $V_s$ = Shell side mass velocity, Kg/m <sup>2</sup> s $N$ = Number of shell passes $\rho_s$ = Density of shell fluid, Kg/m <sup>3</sup> $D_h$ = Hydraulic diameter of the shell $D_i$ = Inside diameter of the shell $d_o$ = Outside diameter of the tube $N_t$ = Number of tubes in the shell
11.	If $C_c < C_h$	$\epsilon = \frac{T_{co} - T_{ci}}{T_{hi} - T_{ci}}$	2.190	
12.	Correction factor	$F = \frac{\sqrt{R^2 + 1}}{R - 1} \ln \left[ \frac{1 - P}{1 - RP} \right]$ $= \frac{\sqrt{R^2 + 1}}{\ln \left[ \frac{(2/P) - 1 - R + \sqrt{R^2 + 1}}{(2/P) - 1 - R - \sqrt{R^2 + 1}} \right]}$	2.191	
		Where, $P = \left[ \frac{T_{co} - T_{ci}}{T_{hi} - T_{ci}} \right]$		
		$R = \left[ \frac{T_{hi} - T_{ho}}{T_{co} - T_{ci}} \right]$		
13.	Tube-side pressure drop	$\Delta P_t = \frac{f V_t^2 L n}{2g \rho_t d_i \phi_t}$ <p>Where, <math>\phi_t = (\mu/\mu_w)^m</math>, <math>m = 0.14</math> Re &gt; 2100  <math>= 0.25</math> Re &lt; 2100</p>	2.192	
14.	Shell-side pressure drop for an un baffled shell	$\Delta P_s = \frac{f V_s^2 L N}{2g \rho_s D_h \phi_s}$ <p>Where, <math>\phi_s = \left[ \frac{\mu}{\mu_w} \right]^{0.14}</math></p> $D_h = \frac{4[(\pi D_i^2/4) - (\pi d_o^2 N_t)]}{\pi d_o N_t + \pi D_i}$ $\Delta P_s = \frac{f V_s^2 D_i (N_h + 1)}{2g \rho_s D_h \phi_s}$	2.193	
15.	Shell-side pressure drop for a shell with segmental baffles	$N_b = \text{Number of bottles}$	2.194	

**Table 2.4** Effectiveness and NTU relations for Heat Exchangers

Sl. No.	Flow Arrangement	Effectiveness Relation ( $\epsilon$ )	NTU Relation ( $N$ )
<i>Concentric tube</i>			
1.	Parallel flow	$\epsilon = \frac{1 - \exp[-N(1+C)]}{1+C}$	$N = \frac{-\ln[1-\epsilon(1+C)]}{1+C}$
2.	Counter flow	$\epsilon = \frac{1 - \exp[-N(1-C)]}{1-C \exp[-N(1-C)]} \quad (C < 1)$	$N = \frac{1}{C-1} \ln \left( \frac{\epsilon-1}{\epsilon C-1} \right) \quad (C < 1)$
		$\epsilon = \frac{N}{1+N} \quad (C = 1)$	$N = \frac{\epsilon}{1-\epsilon} \quad (C = 1)$
3.	<i>Cross flow – Both fluids unmixed</i>	$\epsilon = 1 - \exp \left[ \frac{\exp(-NCn) - 1}{Cn} \right]$ where, $C \times n = N^{-0.22}$ ,	
4.	<i>Cross flow – Both fluids mixed</i>	$\epsilon = \left[ \frac{1}{1 - \exp(-N)} + \frac{C}{1 - \exp(-NC)} - \frac{1}{N} \right]^{-1}$	
5.	$C_{\max}$ mixed, $C_{\min}$ unmixed	$\epsilon = (1/C) \{1 - \exp[-C(1 - e^{-N})]\}$	$N = -\ln \left[ 1 + \frac{1}{C} \ln(1 - C\epsilon) \right]$
6.	$C_{\max}$ unmixed, $C_{\min}$ mixed	$\epsilon = 1 - \exp \{-(1/C) [1 - \exp(-NC)]\}$	$N = \frac{-1}{C} \ln [1 + C \ln(1 - \epsilon)]$
7.	<i>Shell and Tube</i> One shell pass, 2, 4, 6 ... tube passes	$\epsilon = 2 \left\{ 1 + c + (1 + c^2)^{1/2} \times \frac{1 + \exp[-N(1+C)^{1/2}]}{1 - \exp[-N(1+C)^{1/2}]} \right\}^{-1}$	$N = -(1 + C^2)^{-1/2} \ln \left( \frac{E-1}{E+1} \right)$ $E = \frac{2/\epsilon - (1+C)}{(1+C^2)^{1/2}}$

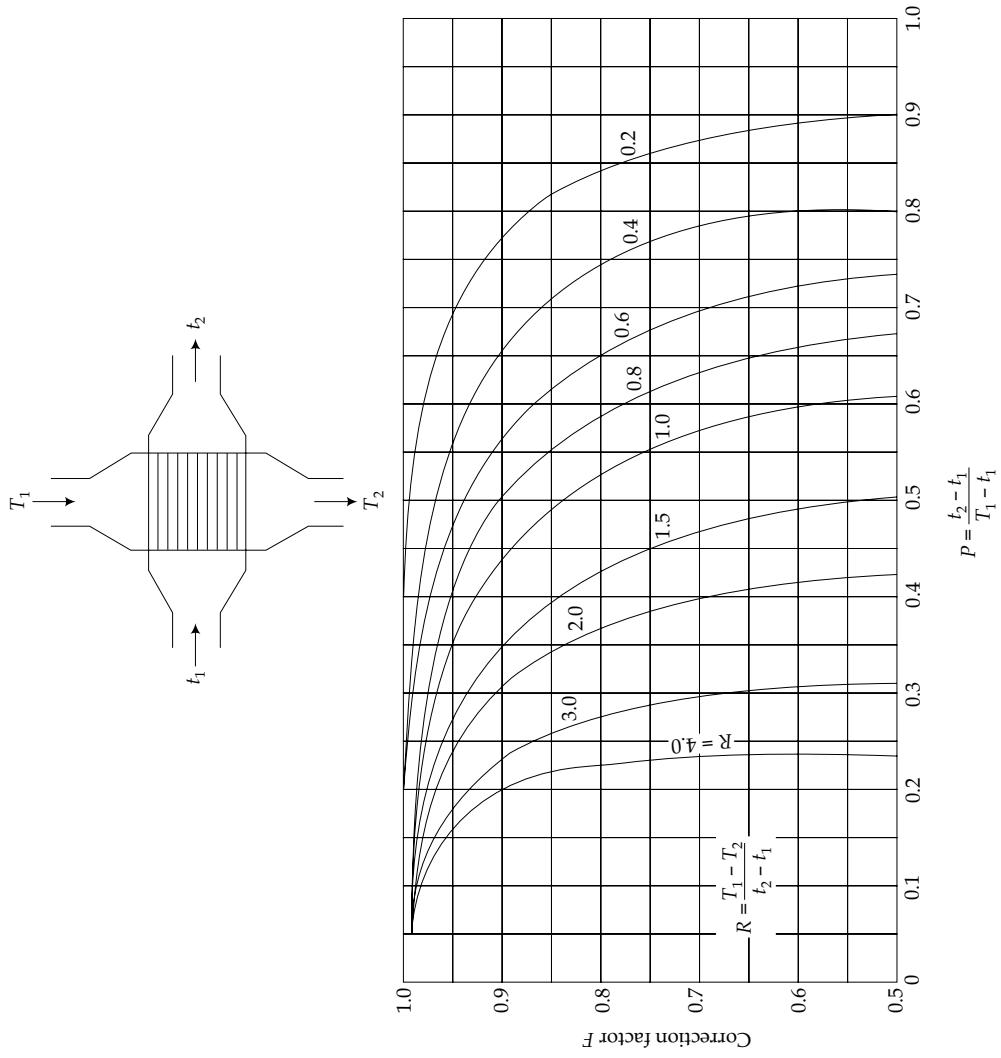


**Fig. 2.13** Correction factor plot for exchanger with one shell pass and two, four, or any multiple of tube passes



**Fig. 2.14** Correction factor plot for exchanger with two shell pass and four, eight, or any multiple tube passes





**Fig. 2.16** Correction factor plot for single pass cross flow heat exchangers, one fluid mixed and the other unmixed

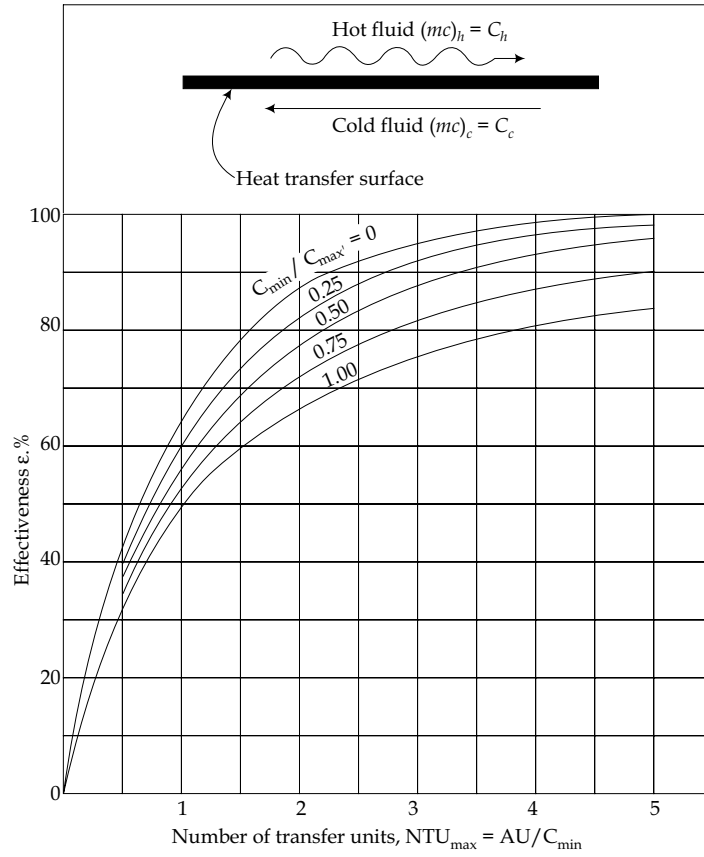
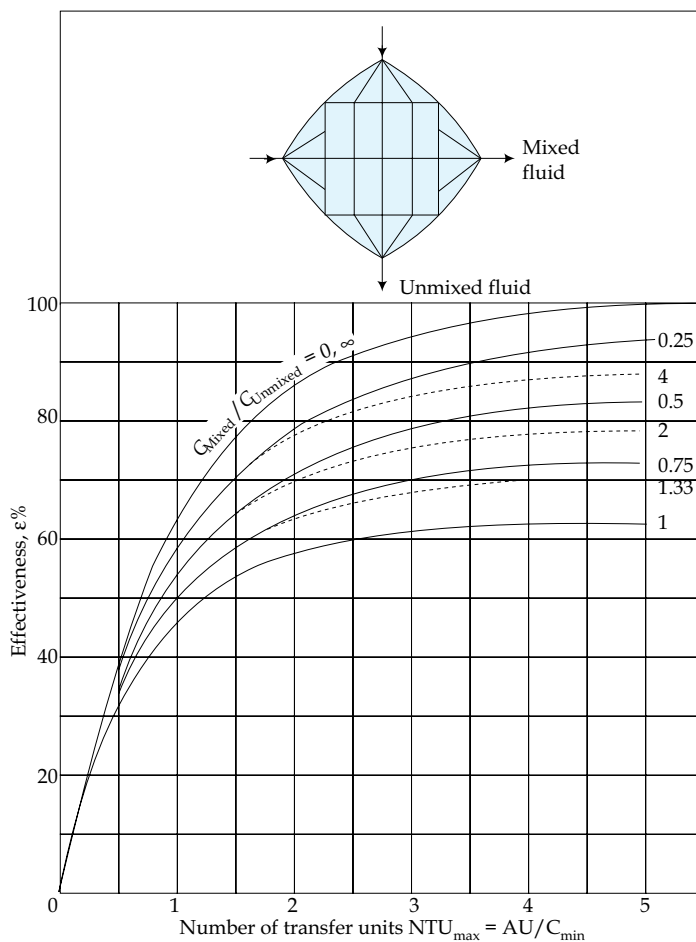
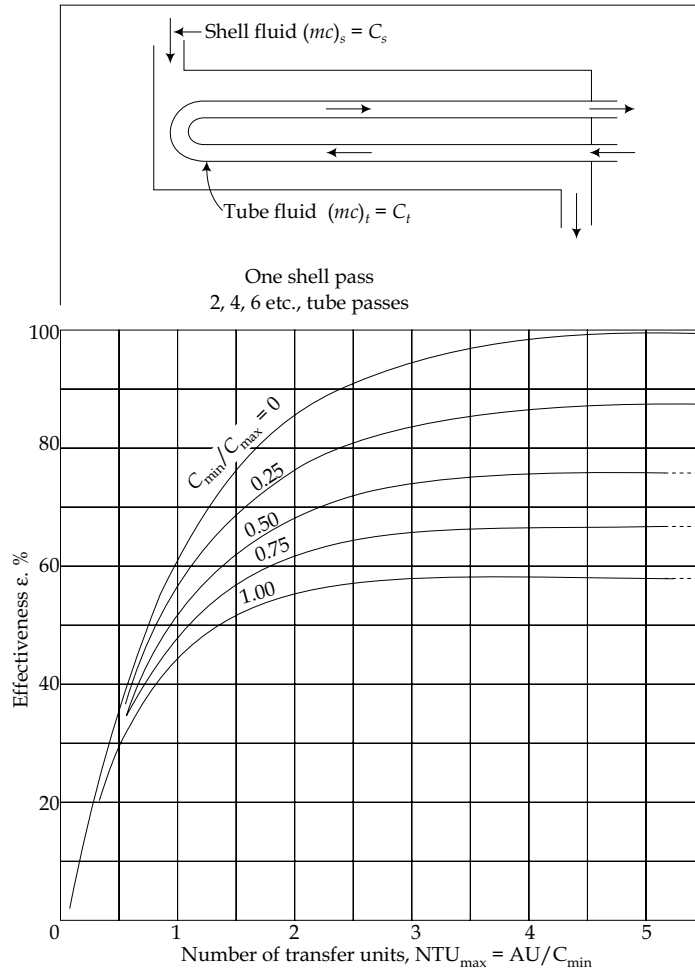


Fig. 2.18 Effectiveness for counter flow exchanger



**Fig. 2.19** Effectiveness for cross flow exchanger with one fluid mixed



**Fig. 2.21** Effectiveness for 1-2 parallel counter flow exchanger performance

**2.3 RADIATION HEAT TRANSFER**

Sl. No.	Description	Correlation	Eq. No.	Notations
1.	Wave length	$\lambda = C/\gamma$	2.195	$\lambda$ = Wavelength, $\mu\text{m}$
2.	Quantum theory	$E = h\gamma$	2.196	$c$ = Speed of light = $C_0/n$
3.	Energy density of radiation per unit volume and per unit wavelength	$U_\lambda = \frac{8\pi hc\lambda^{-5}}{e^{hc/\lambda kT} - 1}$	2.197	$C_0 = 2.998 \times 10^8$ m/s $n = 1$ for gases $n = 1.5$ for Glass
4.	Planck's Equation - manochromatic emissive power of a black body	$E_{b\lambda} = \frac{C_1\lambda^{-5}}{\exp(C_2/\lambda T) - 1}$	2.198	$E$ = Energy radiated $h$ = Plank coefficient = $6.625 \times 10^{-34}$ J-S
5.	Wien's displacement law	$\lambda_{\text{max}} \cdot T = C_3$	2.199	$\gamma$ = Frequency of Quantum
6.	Stefan - Boltzmann law - Emissive power of a black body	$E_b = \sigma T^4$	2.200	$K$ = Boltzmann's constant = $1.38066 \times 10^{-23}$ J/molecule.K
7.	Radiation emitted by a black body per unit area over wavelength band from $\lambda = 0$ to $\lambda$	$E_{b,0-\lambda}(T) = \int_0^\lambda E_{b\lambda}(T) d\lambda$	2.201	$T$ = Absolute temperature
8.	Radiant energy emitted by the black body between wavelengths $\lambda_1$ and $\lambda_2$	$E_{b,\lambda_1-\lambda_2} = E_{b,0-\lambda_2} - E_{b,0-\lambda_1}$	2.202	$C_1 = 3.743 \times 10^8$ W. $\mu\text{m}^4/\text{m}^2$ $C_2 = 1.4387 \times 10^4$ $\mu\text{mK}$ $C_3 = 2897.6$ $\mu\text{mK}$
9.	Radiant property	$\alpha + \tau + \rho = 1$	2.203	$\sigma$ = Stefan-Boltzmann's constant
10.	Emissivity of real surface	$\epsilon = (E/E_b)$	2.204	= $5.67 \times 10^{-8}$ W/ $\text{m}^2\text{K}^4$
11.	Radiant energy emitted by real surface per unit area at temperature $T_r$	$E = \epsilon \cdot \sigma T^4$	2.205	$\lambda_{\text{max}}$ = Wavelength at which emissive power is maximum
12.	Kirchoff's Law	$\epsilon = \alpha$	2.206	$\alpha$ = Absorptivity
13.	Relation between Irradiation (G) and Radiosity (J)	$J = \rho G + \epsilon \cdot E_b$	2.207	$\tau$ = Transmissivity
14.	Net energy leaving a surface	$Q = J - G \text{   } A = \frac{E_b - J}{(1 - \epsilon)} \text{   } \epsilon A$	2.208	$\rho$ = Reflectivity $E$ = Emissive power of a surface
15.	Total radiation which leaves surface 1 that reaches surface 2	$Q_{12} = \frac{J_1 - J_2}{1/A_1 F_{12}}$	2.209	$E_b$ = Emissive power of a black surface $\epsilon$ = Emissivity

16.	Radiation-balance equation for constant surface temperature of $t_{th}$ surface	$J_i [1 - F_{ji}(1 - \epsilon_i)] - (1 - \epsilon_i) \sum_{j \neq i} F_{ij} J_j = \epsilon_i E_{bi}$	2.210	<p><math>A</math> = Heat transfer area, <math>m^2</math>  <math>J_1, J_2</math> = Radiosity of surface 1 &amp; 2 respectively  <math>A_1, A_2</math> = Areas of surface 1 &amp; 2 respectively  <math>J_i</math> = Radiosity of <math>i</math>'th surface  <math>\epsilon_i</math> = Emissivity of <math>i</math>'th surface  <math>F_{ij}</math> = View factor from surface <math>i</math>' to surface <math>j</math>'  <math>E_{bi} = J_i</math> for insulated surface  <math>F_{A_1-A_2}</math> (or) <math>F_{1,2}</math> = View factor from (or) a surface of area <math>A_1</math> to another surface of area <math>A_2</math>  <math>r</math> = Distance between the two surfaces  <math>f_{12}</math> = Interchange factor from surface 1 to surface 2  <math>T_1, T_2</math> = Temperature of body 1 &amp; 2, respectively  <math>Q_{net}</math> = Net heat interchange between the bodies, W</p>
17.	Radiation-balance equation for surface in radiant balance i.e., $Q/A$	$J_i(1 - F_{ji}) - \sum_{j \neq i} F_{ij} J_j = 0$	2.211	
18.	Radiation-balance equation for surface with specified heat flux	$J_i(1 - F_{ji}) - \sum_{j \neq i} F_{ij} J_j = \frac{Q_i}{A_i}$	2.212	
19.	Shape factor (or) view factor between two finite surfaces	$F_{A_1-A_2} = \frac{1}{A_1} \int \int_{A_1, A_2} \frac{\cos \theta_1 \cos \theta_2}{\pi r^2} dA_1 dA_2$	2.213a	
		$F_{A_2-A_1} = \frac{1}{A_2} \int \int_{A_1, A_2} \frac{\cos \theta_1 \cos \theta_2}{\pi r^2} dA_1 dA_2$	2.213b	
20.	Radiant heat exchange between two bodies or surfaces	$A_1 F_{A_1-A_2} = A_2 F_{A_2-A_1}$	2.213c	
		$Q_{net} = f_{12} F_{12} \sigma A_1 [T_1^4 - T_2^4]$ $= f_{21} F_{21} \sigma A_2 [T_1^4 - T_2^4]$	2.214 2.215	
		$Q_N = \frac{A\sigma [T_1^4 - T_2^4]}{(N+1)(2/\epsilon - 1)}$	2.216	
21.	<i>Radiation Shields</i> For Parallel Plate system containing $N$ shields between two surfaces with emissivities of all surfaces equal			
22.	Ratio of heat transfer rates for parallel plate systems with $N$ shields and with no shield	$\frac{Q_N}{Q_0} = \frac{1}{N+1}$ $Q_0 = \frac{A\sigma [T_1^4 - T_2^4]}{(2/\epsilon - 1)}$	2.217	

**Table 2.4a** View factor and Interchange factor for different configurations

Sl. No.	Configuration	View factor ( $F_{12}$ )	Inter change factor ( $f_{12}$ )	Notations
1.	Infinite Parallel Plates	1	$\left( \frac{1}{\epsilon_1} + \frac{1}{\epsilon_2} - 1 \right)$	$\epsilon_1$ & $\epsilon_2$ = emissivity of body 1 and 2, respectively
2.	Body 1 completely enclosed by body 2, body 1 is very small	1	$\epsilon_1$	
3.	Body 1 completely enclosed by body 2, body 1 is very large	1	$1 - \frac{A_1}{A_2} \left( \frac{1}{\epsilon_2} - 1 \right)$	
4.	Concentric spheres or Infinitely long concentric cylinders	1	$1 - \frac{A_1}{A_2} \left( \frac{1}{\epsilon_2} - 1 \right)$	
5.	Two rectangles with common side at right angles to each other	1	$\epsilon_1 \epsilon_2$	
6.	Radiant exchange between two small gray bodies	$F_{12}$	$\epsilon_1 \epsilon_2$	
7.	Radiant exchange between two black surfaces	$F_{12}$	1	

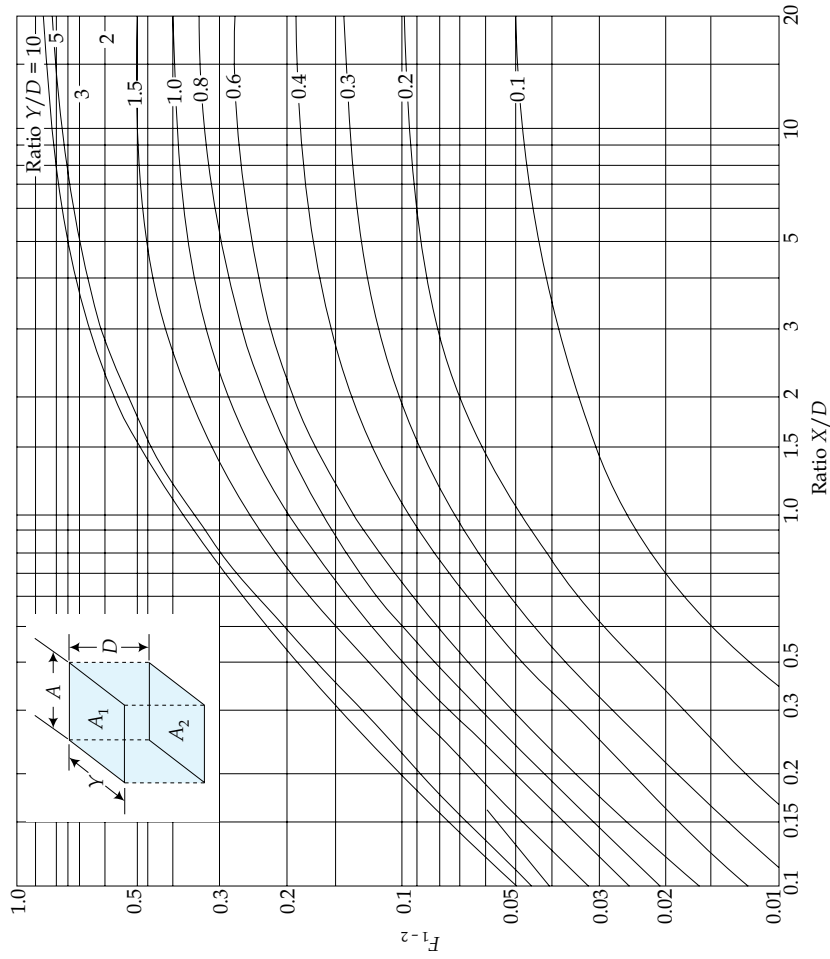
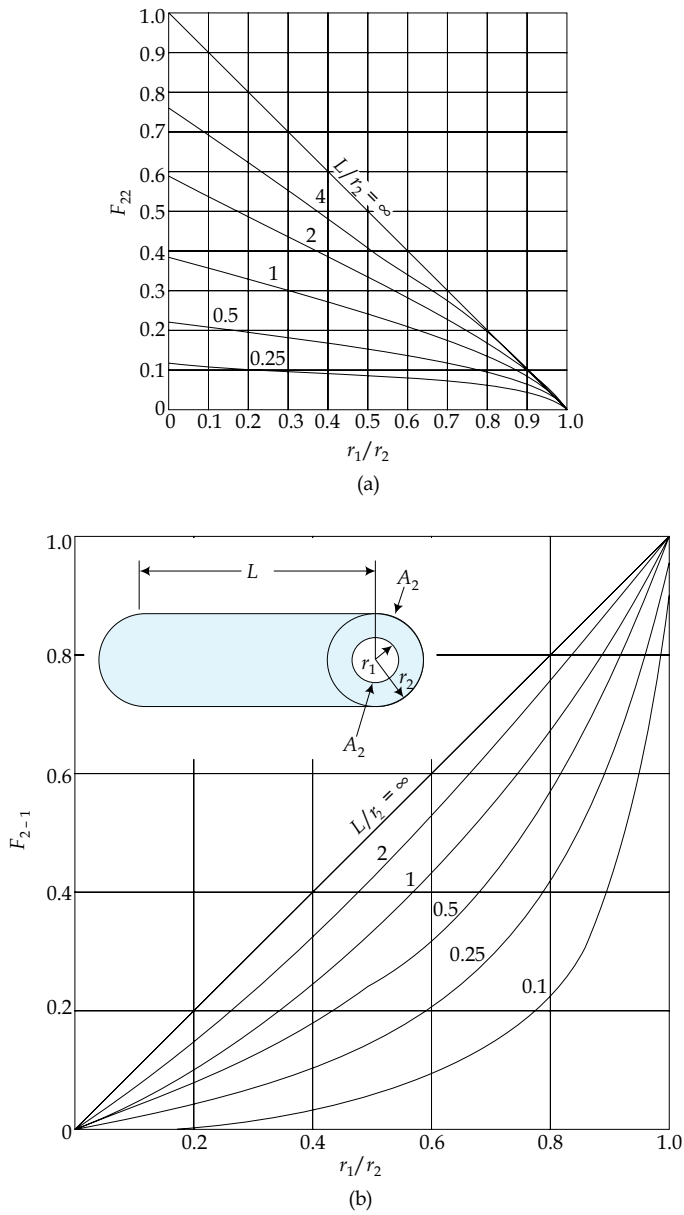
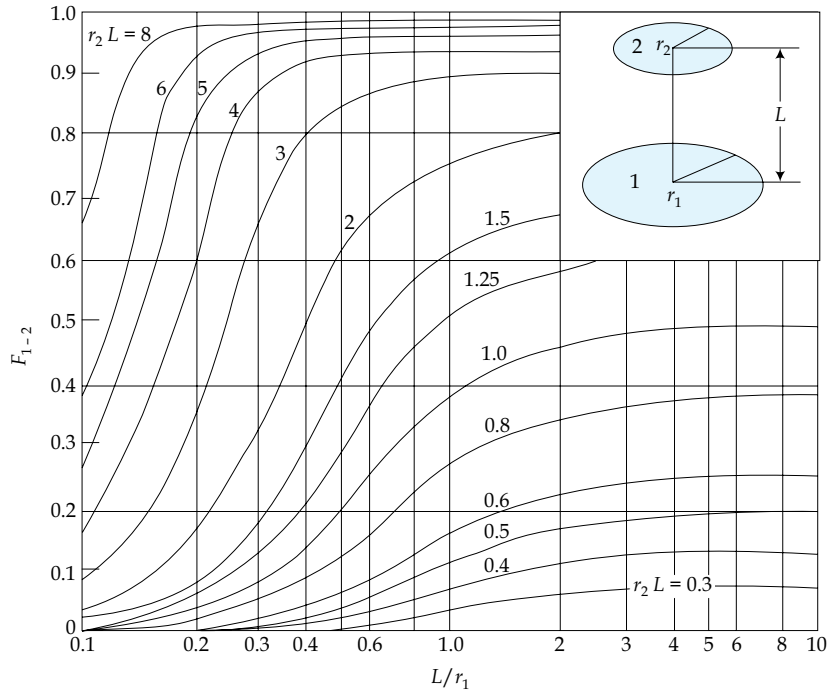


Fig. 2.24 Radiation shape factor for radiation between parallel rectangles





**Fig. 2.27** Radiation shape factor for two concentric cylinders of finite length (a) Outer cylinder to itself (b) Outer cylinder to inner cylinders



**Fig. 2.28** Radiation shape factor for radiation between two parallel concentric disks

9.	Steady state diffusion across a layer of thickness $\Delta x$	$\dot{m}_A = \frac{-DA[C_{A_2} - C_{A_1}]}{\Delta x}$ $= h_D \cdot A (C_{A2} - C_{A1})$	2.227	$D_{\text{mix}}$ = Diffusion coefficient when component A diffuses into a mixture $P_{\text{mix}2}$ & $P_{\text{mix}1}$ = Partial Pressure of components other than diffusing component $\dot{m}_{w0}$ = Mass flux of water $M_w$ = Molecular weight of water $P_{w01}, P_{w02}$ = Partial Pressure of water $(\Delta P)_{LM}$ = Log mean Partial pressures $P_{a1}, P_{a2}$ = Partial Pressure of air $h_D$ = Mass transfer coefficient $B_i$ = Biot number $F_o$ = Fourier number $Sh_x$ = Sher wood number $= \frac{h_D \cdot x}{D}$ $Sc$ = Schmidt number $= \frac{\nu}{D} = \frac{\mu}{\rho D}$ $u_\alpha$ = Free stream velocity $D$ = Diffusion coefficient $h$ = Heat transfer coefficient $L_c$ = Lewis number $= \frac{Sc}{Pr} = \frac{\alpha}{D}$ $\alpha$ = Thermal diffusivity
10.	Steady state diffusion of component A into a stagnant mixture of component B, C, etc	$\frac{\dot{m}_A}{A} = D_{\text{mix}} \cdot \frac{P M_A}{RT[x_2 - x_1]} \ln \frac{P_{\text{mix}2}}{P_{\text{mix}1}}$	2.228	
11.	Diffusion of water vapours through air	$\dot{m}_W = \frac{DA}{RT} \frac{M_w \cdot P}{(x_2 - x_1)} \frac{[P_{W_1} - P_{W_2}]}{(\Delta P)_{LM}}$ $(\Delta P)_{LM} = \frac{P_{a_2} - P_{a_1}}{\ln \frac{P_{a_2}}{P_{a_1}}}$	2.229	
12.	Transient diffusion in an Infinite plate – The diffusion of 'A' of an Infinite plate of thickness 2L of 'B' having an initial concentration of A as $\dot{m}_{A0}$ suddenly exposed to fluid mixture with concentration $\dot{m}_{A\infty}$ .	$\frac{\dot{m}_A - \dot{m}_{A\infty}}{\dot{m}_{A0} - \dot{m}_{A\infty}} = e^{-Bi_x Fo, m}$ $Bi, m = \frac{h_D L}{\rho D}$ $Fo, m = \frac{\epsilon \cdot D}{L^2}$	2.230	
13.	The diffusion of component 'A' in a semi-infinite plate of material 'B', having an initial concentration of A as $\dot{m}_{A0}$ , the slab is suddenly exposed to surface concentration $\dot{m}_{As}$	$\frac{\dot{m}_A - \dot{m}_{As}}{\dot{m}_{A0} - \dot{m}_{As}} = \text{erf} \left[ \frac{x}{z\sqrt{Dt}} \right]$	2.231	
14.	Flow over a Flat plate Laminar flow ( $Re_x < 5 \times 10^5$ )	$Sh_x = 0.332 Re_x^{1/2} Sc^{1/3}$	2.232	
15.	Turbulent flow ( $Re_x > 5 \times 10^5$ )	$\overline{Sh}_x = 0.664 Re_x^{1/2} Sc^{1/3}$	2.233	
16.	For mixed boundary layer condition	$Sh_L = (0.037 Re_L^{4/5} - 870) Sc^{1/3}$	2.234	
17.	Flow over smooth flat plates: Laminar flow	$\frac{h_D}{u_\alpha} Sc^{2/3} = 0.332 Re_x^{-1/2}$	2.235	

18.	Turbulent flow	$\frac{h_D}{u_\alpha} Sc^{2/3} = 0.0296 Re_x^{-1/5}$	2.236	C & $\eta$ are constants taken from the following table
19.	Flow over cylinders and spheres	$Sh_D = C(Re_D)^\eta Sc^{1/3}$	2.237	
20.	Flow through the pipe Laminar flow for uniform wall mass concentration ( $Re_D < 2000$ )	$Sh = 3.66$	2.238	
21.	Laminar flow for uniform wall mass flux ( $Re_D < 2000$ )	$Sh = 4.36$	2.239	
22.	Turbulent flow ( $Re_D > 2000$ )	$Sh = 0.023 Re^{0.83} Sc^{1/3}$	2.240	
23.	Simultaneous heat and mass transfer	$\frac{h}{h_D} = \rho C_p Le^{2/3}$	2.241	

C &  $\eta$  are constants taken from the following table

Re	C	$\eta$
0.4-4	0.989	0.330
4-40	0.911	0.385
40-4000	0.683	0.466
4,000-40,000	0.193	0.618
40,000-4,00,000	0.027	0.805



Table 2.6 Properties of Non-metals\*\*

Substance	Temperature °C	$K$ $W/m\ ^\circ C$	$\rho$ $kg/m^3$	$C_p$ $kJ/kg.\ ^\circ C$	$\alpha \times 10^7$ $m^2/s$
<i>Structural and heat-resistant materials</i>					
Asphalt	20-55	0.74-0.76			
Brick: Building brick, common	20	0.69	1600	0.84	5.2
Face		1.32	2000		
Carborundum brick	600 1400		18.5 11.1		
Chrome brick	200 550 900	2.32 2.47 1.99	3000	0.84	9.2 9.8 7.9
Diatomaceous earth, molded and fired	200 870	0.24 0.31			
Fireclay brick, burnt 2426°F	500 800 1100	1.04 1.07 1.09	2000	0.96	5.4
Burnt 2642°F	500 800 1100	1.28 1.37 1.40	2300	0.96	5.8
Missouri	200 600 1400	1.00 1.47 1.77	2600	0.96	4.0
Magnesite	200 650 1200	3.81 2.77 1.90		1.13	
Cement portland Mortar	23	0.29 1.16	1500		
Concrete, cinder	23	0.76			
Stone, 1-2-4 mix	20	1.37	1900-2300	0.88	8.2-6.8
Glass, window	20	0.78 (avg)	2700	0.84	3.4
Corosilicate	30-75	1.09	2200		
Plaster, gypsum	20	0.48	1440	0.84	4.0
Metal lath	20	0.47			
Wood lath	20	0.28			
Stone: Granite	100-300	1.73-3.98	2640	0.82	8-18
Limestone		1.26-1.33	2500	0.90	5.6-5.9
Marble		2.07-2.94	2500-2700	0.80	10-13.6
Sandstone		1.83	2160-2300	0.71	11.2-11.9

Wood (across the grain):					
balsa, 8.8 lb/ft <sup>3</sup>	30	0.055	140		
Cypress	30	0.097	460		
Fir	23	0.11	420	2.72	0.96
Maple or oak	30	0.166	540	2.4	1.28
Yellow pine	23	0.147	640	2.8	0.82
White pine	30	0.112	430		
<i>Insulating material</i>					
Asbestos: Loosely packed	-45	0.149			
	0	0.154	470-570	0.816	3.3-4
	100	0.161			
Asbestos-cement	20	0.74			
Boards					
Sheets Felt,	51	0.166			
40 laminations/in	38	0.057			
	150	0.069			
	260	0.083			
20 laminations/in	38	0.078			
	150	0.095			
	260	0.112			
corrugated, 4 plies/in	38	0.087			
	93	0.100			
	150	0.119			
Asbestos cement	-	2.08			
Balsam wool, 2.2 lb/ft <sup>3</sup>	32	0.04	35		
Cardboard, corrugated	-	0.064			
Celotex	32	0.048			
Corkboard, 10 lb/ft <sup>3</sup>	30	0.043	160		
Cork, regranulated	32	0.045	45-120	1.88	2-5.3
Ground	32	0.043	150		
Diatomaceous earth (sio-o-cel)	0	0.061	320		
Felt, hair	30	0.036	130-200		
Wool	30	0.052	330		
Fiber, insulating board	20	0.048	240		
Glass wool, 1.5 lb/ft <sup>3</sup>	23	0.038	24	0.7	22.6
Insulex, dry	32	0.064			
		0.144			
Kapok	30	0.035			
Magnesia, 85%	38	0.067	270		
	93	0.071			
	150	0.074			
	204	0.080			

0	1,438.46	1.3636	0.257	0.211	1.081	2.38	$1.94 \times 10^{-3}$
10	1,412.51	1.3645	0.232	0.204	1.066	2.18	
20	1,386.40	1.3653	0.210	0.199	1.050	2.00	
30	1,359.33	1.3662	0.190	0.192	1.035	1.83	
40	1,329.22	1.3674	0.173	0.185	1.019	1.70	
50	1,299.10	1.3683	0.162	0.177	0.999	1.61	
<i>Dichlorodifluoromethane (Freon), CCl<sub>2</sub>F<sub>2</sub></i>							
-50	1,546.75	0.8750	$0.310 \times 10^{-6}$	0.067	$0.501 \times 10^{-7}$	6.2	$2.63 \times 10^{-3}$
-40	1,518.71	0.8847	0.279	0.069	0.514	5.4	
-30	1,489.56	0.8956	0.253	0.069	0.526	4.8	
-20	1,460.57	0.9073	0.235	0.071	0.539	4.4	
-10	1,429.49	0.9203	0.221	0.073	0.550	4.0	
0	1,397.45	0.9345	$0.214 \times 10^{-6}$	0.073	$0.557 \times 10^{-6}$	3.8	
10	1,364.30	0.9496	0.203	0.073	0.560	3.6	
20	1,330.18	0.9659	0.198	0.073	0.560	3.5	
30	1,295.10	0.9835	0.194	0.071	0.560	3.5	
40	1,257.13	1.0019	0.191	0.069	0.555	3.5	
50	1,215.96	1.0216	0.190	0.067	0.545	3.5	
<i>Glycerin, C<sub>3</sub>H<sub>5</sub>(OH)<sub>3</sub></i>							
0	1,276.03	2.261	0.00831	0.282	$0.983 \times 10^{-7}$	$4.7 \times 10^3$	$0.50 \times 10^{-3}$
10	1,270.11	2.319	0.00300	0.284	0.965	1.0	
20	1,264.02	2.386	0.00118	0.286	0.947	2.5	
30	1,258.09	2.445	0.00050	0.286	0.929	5.38	
40	1,252.01	2.512	0.00022	0.286	0.914	2.45	
50	1,244.96	2.583	0.00015	0.287	0.893	1.63	
<i>Ethylene glycol, C<sub>2</sub>H<sub>4</sub>(OH)<sub>2</sub></i>							
0	1,130.75	2.294	$57.53 \times 10^{-6}$	0.242	$0.934 \times 10^{-7}$	615	$0.65 \times 10^{-3}$
20	1,116.65	2.382	19.18	0.249	0.939	204	
40	1,101.43	2.474	8.69	0.256	0.939	93	
60	1,087.66	2.562	4.75	0.260	0.932	51	
80	1,077.56	2.650	2.98	0.261	0.921	32.4	
100	1,058.50	2.742	2.03	0.263	0.908	22.4	
<i>Engine oil (unused)</i>							
0	899.12	1.796	0.00428	0.147	$0.911 \times 10^{-7}$	47,100	$0.70 \times 10^{-3}$
20	888.23	1.880	0.00090	0.145	0.872	10,400	
40	876.05	1.964	0.00024	0.144	0.834	2,870	
60	864.04	2.047	$0.834 \times 10^{-4}$	0.140	0.800	1,050	
80	852.02	2.131	0.375	0.138	0.769	490	
100	840.01	2.219	0.203	0.137	0.738	276	



120	828.96	2.307	0.124	0.135	0.710	175	
140	816.94	2.395	0.080	0.133	0.686	116	
160	805.89	2.483	0.056	0.132	0.663	84	
<i>Mercury, Hg</i>							
0	13,628.22	0.1403	$0.124 \times 10^{-6}$	8.20	$42.99 \times 10^7$	0.0288	$1.82 \times 10^{-4}$
20	13,579.04	0.1394	0.114	8.69	46.06	0.0249	
50	13,505.84	0.1386	0.104	9.40	50.22	0.0207	
100	13,384.58	0.1373	0.0928	10.51	57.16	0.0162	
150	13,264.28	0.1365	0.0853	11.49	63.54	0.0134	
200	13,144.94	0.1570	0.0802	12.34	69.08	0.0116	
250	13,025.60	0.1357	0.0765	13.07	74.06	0.0103	
315.5	12,847	0.134	0.0673	14.02	81.5	0.0083	
<i>Eutectic calcium chloride solution, 29.9% CaCl<sub>2</sub></i>							
-50	1,319.76	2.608	$36.35 \times 10^{-6}$	0.402	1.166	312	
-40	1,314.96	2.6356	24.97	0.415	1.200	208	
-30	1,310.15	2.6611	17.18	0.429	1.234	139	
-20	1,305.51	2.688	11.04	0.445	1.267	87.1	
-10	1,300.70	2.713	6.96	0.459	1.300	53.6	
0	1,296.06	2.738	4.39	0.472	1.332	33.0	
10	1,291.41	2.763	3.35	0.485	1.363	24.6	
20	1,286.61	2.788	2.72	0.498	1.394	19.6	
30	1,281.96	2.814	2.27	0.511	1.419	16.0	
40	1,277.16	2.839	1.92	0.523	1.445	13.3	
50	1,272.51	2.868	1.65	0.535	1.468	11.3	
<i>Methyl chloride, CH<sub>3</sub>Cl</i>							
-50	1,052.58	1.4759	$0.320 \times 10^{-6}$	0.215	1.388	2.31	
-40	1,033.35	1.4826	0.318	0.209	1.368	2.32	
-30	1,016.53	1.4922	0.314	0.202	1.337	2.35	
-20	999.39	1.5043	0.309	0.196	1.301	2.38	
-10	981.45	1.5194	0.306	0.187	1.257	2.43	
0	962.39	1.5378	0.302	0.178	1.213	2.49	
10	942.36	1.5600	0.297	0.171	1.166	2.55	
20	923.31	1.5860	0.293	0.163	1.112	2.63	
30	903.12	1.6161	0.288	0.154	1.058	2.72	
40	883.10	1.6504	0.281	0.144	0.996	2.83	
50	861.15	1.6890	0.274	0.133	0.921	2.97	

<sup>†</sup>Constructed based on the data compiled in Refs. [ 23,24 ]

**Table 2.8** Properties of Water (Saturated Liquid)\*\*

$^{\circ}\text{C}$	$c_p$ kJ/kg. $^{\circ}\text{C}$	$\rho$ , kg/m <sup>2</sup>	$\mu$ , kg/ms	$k$ , W/m $^{\circ}\text{C}$	$Pr$	$Gr_x Pr$
0	4.225	999.8	$1.79 \times 10^{-3}$	0.566	13.25	
4.44	4.208	999.8	1.55	0.575	11.35	$1.91 \times 10^9$
10	4.195	999.2	1.31	0.585	9.40	$6.34 \times 10^9$
15.56	4.186	998.6	1.12	0.595	7.88	$1.08 \times 10^{10}$
21.11	4.179	997.4	$9.8 \times 10^{-4}$	0.604	6.78	$1.46 \times 10^{10}$
26.67	4.179	995.8	8.6	0.614	5.85	$1.91 \times 10^{10}$
32.22	4.174	994.9	7.65	0.623	5.12	$2.48 \times 10^{10}$
37.78	4.174	993.0	6.82	0.630	4.53	$3.3 \times 10^{10}$
43.33	4.174	990.6	6.16	0.637	4.04	$4.19 \times 10^{10}$
48.89	4.174	988.8	5.62	0.644	3.64	$4.89 \times 10^{10}$
54.44	4.179	985.7	5.13	0.649	3.30	$5.66 \times 10^{10}$
60	4.179	983.3	4.71	0.654	3.01	$6.48 \times 10^{10}$
65.55	4.183	980.3	4.3	0.659	2.73	$7.62 \times 10^{10}$
71.1	4.186	977.3	4.01	0.665	2.53	$8.84 \times 10^{10}$
76.67	4.191	973.7	3.72	0.668	2.33	$9.85 \times 10^{10}$
82.22	4.195	970.2	3.47	0.673	2.16	$1.09 \times 10^{11}$
87.78	4.199	966.7	3.27	0.675	2.03	
93.33	4.204	963.2	3.06	0.678	1.90	
104.4	4.216	955.1	2.67	0.684	1.66	
115.6	4.229	946.7	2.44	0.685	1.51	
126.7	4.250	937.2	2.19	0.685	1.36	
137.8	4.271	928.1	1.98	0.685	1.24	
148.9	4.296	918.0	1.86	0.684	1.17	
176.7	4.371	890.4	1.57	0.677	1.02	
204.4	4.467	859.4	1.36	0.665	1.00	
232.2	4.585	825.7	1.20	0.646	0.85	
260	4.731	785.2	1.07	0.616	0.83	
287.7	5.024	735.5	$9.51 \times 10^{-5}$			
315.6	5.703	678.7	8.68			

**Table 2.10** Properties of Gases at Atmospheric Pressure\*\*

$T, K$	$\rho, \text{kg/m}^3$	$c_p, \text{kJ/kg} \cdot ^\circ\text{C}$	$\mu, \text{kg/m} \cdot \text{s}$	$V, \text{m}^2/\text{s}$	$k, \text{W/m} \cdot ^\circ\text{C}$	$\alpha, \text{m}^2/\text{s}$	$Pr$
<i>Helium</i>							
144	0.3379	5.200	$125.5 \times 10^{-7}$	$37.11 \times 10^{-6}$	0.0928	$0.5275 \times 10^{-4}$	0.70
200	0.2435	5.200	156.6	64.38	0.1177	0.9288	0.694
255	0.1906	5.200	181.7	95.50	0.1357	1.3675	0.70
366	0.13280	5.200	230.5	173.6	0.1691	2.449	0.71
477	0.10204	5.200	275.0	269.3	0.197	3.716	0.72
589	0.08282	5.200	311.3	375.8	0.225	5.215	0.72
700	0.07032	5.200	347.5	494.2	0.251	6.661	0.72
800	0.06023	5.200	381.7	634.1	0.275	8.774	0.72
<i>Hydrogen</i>							
150	0.16371	12.602	$5.595 \times 10^{-7}$	$34.18 \times 10^{-6}$	0.0981	$0.475 \times 10^{-4}$	0.718
200	0.12270	13.540	6.813	55.53	0.1282	0.772	0.719
250	0.09819	14.059	7.919	80.64	0.1561	1.130	0.713
300	0.08185	14.314	8.963	109.5	0.182	1.554	0.706
350	0.07016	14.436	9.954	141.9	0.206	2.031	0.697
400	0.06135	14.491	10.864	177.1	0.228	2.568	0.690
450	0.05462	14.499	11.779	215.6	0.251	3.164	0.682
500	0.04918	14.507	12.636	257.0	0.272	3.817	0.675
550	0.04469	14.532	13.475	301.6	0.292	4.516	0.668
600	0.04085	14.537	14.285	349.7	0.315	5.306	0.664
700	0.03492	14.574	15.89	455.1	0.351	6.903	0.659
800	0.03060	14.675	17.40	569	0.384	8.563	0.664
900	0.02723	14.821	18.78	690	0.412	10.217	0.676
<i>Oxygen</i>							
150	2.6190	0.9178	$11.490 \times 10^{-6}$	$4.387 \times 10^{-6}$	0.01367	$0.05688 \times 10^{-4}$	0.773
200	1.9559	0.9131	14.850	7.593	0.01824	0.10214	0.745
250	1.5618	0.9157	17.87	11.45	0.02259	0.15794	0.725
300	1.3007	0.9203	20.63	15.86	0.02676	0.22353	0.709
350	1.1133	0.9291	23.16	20.80	0.03070	0.2968	0.702
400	0.9755	0.9420	25.54	26.18	0.03461	0.3768	0.695
450	0.8682	0.9567	27.77	31.99	0.03828	0.4609	0.694
500	0.7801	0.9722	29.91	38.34	0.04173	0.5502	0.697
550	0.7096	0.9881	31.97	45.05	0.04517	0.641	0.700
<i>Nitrogen</i>							
200	1.7108	1.0429	$12.947 \times 10^{-6}$	$7.568 \times 10^{-6}$	0.01824	$0.10224 \times 10^{-4}$	0.747
300	1.1421	1.0408	17.84	15.63	0.02620	0.22044	0.713

400	0.8538	1.0459	21.98	25.74	0.03335	0.3734	0.691
500	0.6824	1.0555	25.70	37.66	0.03984	0.5530	0.684
600	0.5687	1.0756	29.11	51.19	0.04580	0.7486	0.686
700	0.4934	1.0969	32.13	65.13	0.05123	0.9466	0.691
800	0.4277	1.1225	34.84	81.46	0.05609	1.1685	0.700
900	0.3796	1.1464	37.49	91.06	0.06070	1.3946	0.711
1000	0.3412	1.1677	40.00	117.2	0.06475	1.6250	0.724
1100	0.3108	1.1857	42.28	136.0	0.06850	1.8591	0.736
1200	0.2851	1.2037	44.50	156.1	0.07184	2.0932	0.748
<i>Carbon dioxide</i>							
220	2.4733	0.783	$11.105 \times 10^{-6}$	$4.490 \times 10^{-6}$	0.010805	$0.05920 \times 10^{-4}$	0.818
250	12.1657	0.804	12.590	5.813	0.012884	0.07401	0.793
300	1.7973	0.871	14.958	8.321	0.016572	0.10588	0.770
350	1.5362	0.900	17.205	11.19	0.02047	0.14808	0.755
400	1.3424	0.942	19.32	14.39	0.02461	0.19463	0.738
450	1.1918	0.980	21.34	17.90	0.02897	0.24813	0.721
500	1.0732	1.013	23.26	21.67	0.03352	0.3084	0.702
550	0.9739	1.047	25.08	25.74	0.03821	0.3750	0.685
600	0.8938	1.076	26.83	30.02	0.04311	0.4483	0.668
<i>Ammonia, NH<sub>3</sub></i>							
273	0.7929	2.177	$9.353 \times 10^{-6}$	$1.18 \times 10^{-5}$	0.0220	$0.1308 \times 10^{-4}$	0.90
323	0.6487	2.177	11.035	1.70	0.0270	0.1920	0.88
373	0.5590	2.236	12.886	2.30	0.0327	0.2619	0.87
423	0.4934	2.315	14.672	2.97	0.0391	0.3432	0.87
473	0.4405	2.395	16.49	3.74	0.0467	0.4421	0.84
<i>Water vapour</i>							
380	0.5863	2.060	$12.71 \times 10^{-6}$	$2.16 \times 10^{-5}$	0.0246	$0.2036 \times 10^{-4}$	1.060
400	0.5542	2.014	13.44	2.42	0.0261	0.2338	1.040
450	0.4902	1.980	15.25	3.11	0.0299	0.307	1.010
500	0.4405	1.985	17.04	3.86	0.0339	0.387	0.996
550	0.4005	1.997	18.84	4.70	0.0379	0.475	0.991
600	0.3652	2.026	20.67	5.66	0.0422	0.573	0.986
650	0.3380	2.056	22.47	6.64	0.0464	0.666	0.995
700	0.3140	2.085	24.26	7.72	0.0505	0.772	1.000
750	0.2931	2.119	26.04	8.88	0.0549	0.883	1.005
800	0.2739	2.152	27.86	10.20	0.0592	1.001	1.010
850	0.2579	2.186	29.69	11.52	0.0637	1.130	1.019

<i>Carbon monoxide<sup>T</sup></i>							
220	1.55363	1.0429	$13.832 \times 10^{-6}$	$8.903 \times 10^{-6}$	0.01906	$0.1176 \times 10^{-4}$	0.758
250	1.3649	1.0425	15.40	11.28	0.02144	0.15063	0.750
300	1.13876	1.0421	17.843	15.67	0.02525	0.21280	0.737
350	0.97425	1.0434	20.09	20.62	0.02883	0.2836	0.728
400	0.85363	1.0484	22.19	25.99	0.03226	0.3605	0.722
450	0.75848	1.0551	24.18	31.88	0.0436	0.4439	0.718
500	0.68223	1.0635	26.06	38.19	0.03863	0.5324	0.718
550	0.62024	1.0756	27.89	44.97	0.04162	0.6240	0.721
600	0.56850	1.0877	29.60	52.06	0.04446	0.7190	0.724
<i>Fuel Gases<sup>T</sup></i>							
273	1.295	1.043	$15.78 \times 10^{-6}$	$12.20 \times 10^{-6}$	$22.79 \times 10^{-3}$	$0.1688 \times 10^{-4}$	0.72
373	0.950	1.068	20.38	21.54	31.28	0.30833	0.69
473	0.748	1.097	24.49	32.80	40.12	0.48888	0.67
573	0.617	1.122	28.22	45.81	48.38	0.70000	0.65
673	0.525	1.151	31.68	60.38	56.99	0.94166	0.64
773	0.457	1.185	34.84	76.30	65.59	1.21111	0.63
873	0.405	1.214	37.85	93.61	74.19	1.51666	0.62
973	0.363	1.239	40.68	112.1	82.69	1.83888	0.61
1073	0.329	1.264	43.37	131.8	91.53	2.19722	0.60
1173	0.301	1.290	45.9	152.5	100.13	2.58055	0.59
1273	0.275	1.306	48.35	174.3	108.97	3.03333	0.58
1373	0.257	1.323	50.69	197.1	117.46	3.45555	0.57
1473	0.240	1.340	52.98	221.00	126.77	3.9525	0.56
<i>Sulphur dioxide<sup>T</sup></i>							
273	2.926	0.607	$1.23 \times 10^{-6}$	4.14	$8.37 \times 10^{-3}$	$0.0472 \times 10^{-4}$	0.874
373	2.140	0.662	1.64	7.51	12.33	0.0872	0.863
473	1.690	0.712	2.04	11.80	16.63	0.1244	0.856
573	1.395	0.754	2.43	17.10	21.28	0.2014	0.848
673	1.185	0.783	2.81	23.30	25.82	0.2777	0.834
753	1.033	0.808	3.19	30.40	30.70	0.3666	0.822
853	0.916	0.825	3.57	38.30	35.80	0.4722	0.806
953	0.892	0.837	3.94	46.80	41.05	0.5972	0.788
1073	0.743	0.850	4.30	56.50	46.29	0.7333	0.774
1173	0.681	0.858	4.66	66.80	51.87	0.8888	0.755
1273	0.626	0.867	5.02	78.30	57.57	1.0611	0.740

<sup>T</sup>Constructed based on the data compiled in Refs. [18, 23]

Table 2.12 Physical properties of liquid metals\*\*

Metal	Melting point, °C	Boiling point, °C	T, °C	$\rho$ , kg/m <sup>3</sup>	$C_p$ , kJ/(kg·°C)	$\mu \times 10^4$ , kg/(m·s)	$\nu \times 10^6$ , m <sup>2</sup> /s	$k$ , W/(m·°C)	$\alpha \times 10^6$ , m <sup>2</sup> /s	Pr
Bismuth	271	1477	315	10,011	0.144	16.2	0.160	16.4	11.25	0.0142
			538	9,739	0.155	11.0	0.113	0.113	15.6	10.34
Lead	327	1737	760	9,467	0.165	7.9	0.083	15.6	9.98	0.0083
			371	10,540	0.159	2.40	0.023	0.023	16.1	9.61
Lithium	179	1317	704	10,140	0.155	1.37	0.014	14.9	9.48	0.0143
			204.4	509.2	4.365	5.416	1.1098	46.37	0.051	20.96
Mercury	-38.9	357	315.6	498.8	4.270	4.465	0.8982	43.08	20.32	0.0432
			426.7	489.1	4.211	3.927	0.8053	38.24	18.65	15.40
Sodium	97.8	883	537.8	476.3	4.171	3.473	0.7304	30.45	5.038	0.0266
			760.0	13,707.1	0.1415	18.334	0.1342	9.76	10.51	5.716
Potassium	63.9	760	100	13,384.5	0.1373	12.420	0.0928	12.34	6.908	0.0116
			200	13,144.9	0.1570	10.541	0.0802	12.34	12.34	6.908
NaK (56% Na, 44% K)	-11.1	784	93.3	931.6	1.384	7.131	0.7689	84.96	56.29	0.0116
			204.4	907.5	1.339	4.521	0.5010	80.81	66.80	0.0075
NaK (56% Na, 44% K)	-11.1	784	315.6	878.5	1.304	3.294	0.3766	75.78	66.47	0.00567
			426.7	852.8	1.277	2.522	0.2968	69.39	64.05	0.00464
NaK (56% Na, 44% K)	-11.1	784	537.8	823.8	1.264	2.315	0.2821	64.37	62.09	0.00455
			760.0	790.0	1.261	1.964	0.2496	60.56	61.10	59.86
NaK (56% Na, 44% K)	-11.1	784	760.0	767.5	1.270	1.716	0.2245	56.58	58.34	0.00385
			426.7	741.7	0.766	2.108	0.2839	39.45	69.74	0.0041
NaK (56% Na, 44% K)	-11.1	784	537.8	714.4	0.762	1.711	0.2400	36.51	67.39	0.0036
			648.9	690.3	0.766	1.463	0.2116	33.74	64.10	64.10
NaK (56% Na, 44% K)	-11.1	784	760.0	667.7	0.783	1.331	0.1987	31.15	59.86	0.0033
			93.3	889.8	1.130	5.622	0.6347	25.78	27.76	27.76
NaK (56% Na, 44% K)	-11.1	784	204.4	865.6	1.089	3.803	0.4414	26.47	28.23	0.0155
			315.6	838.3	1.068	2.935	0.3515	27.17	30.50	30.50
NaK (56% Na, 44% K)	-11.1	784	426.7	814.2	1.051	2.150	0.2652	27.68	32.52	0.0081
			537.8	788.4	1.047	2.026	0.2581	27.68	33.71	33.71
NaK (56% Na, 44% K)	-11.1	784	648.9	759.5	1.051	1.695	0.2240	27.68	34.86	0.0064

\*\*Data collected from Ref. [23]

**Table 2.13** Normal Fouling Factors\*\*

Type of fluid	Fouling factor, $m^2C/W$
Sea water below 50°C	0.0001
Above 50°C	0.0022
Treated boiler feed water above 50°C	0.0002
River water below 50°C	0.0002 – 0.001
Fuel oil	0.0009
Refrigerating liquid	0.0002
Industrial air	0.0004
Steam, non-oil bearing	0.00009
Steam, oil bearing	0.0003 – 0.0004
Alcohol vapours	0.00009
Quenching oil	0.0007
Organic vapours	0.00009
Organic liquids	0.00018
Brine (Cooling)	0.00018
Exhaust steam	0.00018
Liquid gasoline and liquefied petroleum gases	0.0002 – 0.0004
Vegetable oil	0.0005
Caustic solution	0.0004
Methanol, ethanol and ethylene glycol solutions	0.0004
Natural gas	0.0002 – 0.0004
Compressed air	0.0002
Solvent vapours	0.0004 – 0.0005

\*\*Constructed based on data compiled in Refs. [15 – 24]

**Table 2.14** Magnitude of the Overall heat transfer coefficients\*\*

Sl. No	Fluid Combination	$U, W/m^2 K$
1.	Water to Water	850 – 1700
2.	Water to Oil	110 – 350
3.	Steam condenser (Water in tubes)	1000 – 6000
4.	Ammonia condenser (Water in tubes)	800 – 1400
5.	Alcohol condenser (Water in tubes)	250 – 700
6.	Finned tube heat exchanger (Water in tubes, air in cross flow)	25 – 50
7.	Finned tube heat exchanger (Steam in tubes, air in cross flow)	28 – 280
8.	Gas to Gas	10 – 40
9.	Gas to Water	10 – 250
10.	Light organics to Water	370 – 730
11.	Heavy organics to Water	25 – 370
12.	Steam to Water	1000 – 3500

13.	Steam to ammonia	1000 - 3500
14.	Steam to gases	25 - 250
15.	Steam to light organics	500 - 1000
16.	Steam to heavy organics	30 - 300
17.	Steam to fuel oil	56 - 340
18.	Feed water heater	1100 - 8500
19.	Freon-12 condenser with water coolant	280 - 850

Constructed based on data compiled in Refs. [15 - 24]

**Table 2.15** Surface Tension and other physical properties\*\* for Boiling and Condensation Heat Transfer

Fluid	$T_{Sat}$ K	$P$ ( $10^5$ N/m <sup>2</sup> )	$\rho_l$ (kg/m <sup>3</sup> )	$\rho_v$ (kg/m <sup>3</sup> )	$h_{fg}$ (kJ/kg)	$\sigma$ (N/m)
Ammonia	223	0.409	702	0.38	1417	0.038
	300	10.66	600	8.39	1158	0.020
Ethanol	351	1.013	757	1.44	846	0.018
Helium	4.2	1.013	125	16.9	20.42	0.0001
Hydrogen	20.3	1.013	70.8	—	442	0.002
Lithium	600	$4.2 \times 10^{-9}$	503	—	22340	0.375
	800	$9.6 \times 10^{-9}$	483	$10^{-6}$	21988	0.348
Mercury	630	1.013	12740	3.9	301	0.417
Nitrogen	77.3	1.013	809	4.61	198.4	0.0089
Oxygen	90.2	1.013	1134	—	213.1	0.013
Potassium	400	$1.84 \times 10^{-7}$	814	$2.2 \times 10^{-7}$	2196	0.110
	800	0.0612	720	0.037	2042	0.083
Refrigerant-12	243	1.004	1488	6.27	165.3	0.016
Refrigerant-22	200	0.166	1497	0.87	252.8	0.024
	250	2.174	1360	9.64	221.9	0.016
	300	10.96	1187	46.55	180.1	0.007
Sodium	500	$7.64 \times 10^{-7}$	898	$4.3 \times 10^{-7}$	4438	0.175
	1000	0.1955	776	0.059	4022	0.130
Water	323	0.1235	998	0.08	2383	0.068
	373	1.0133	958	0.6	2257	0.059
	423	4.758	917	2.55	2114	0.048
	473	15.54	865	7.85	1941	0.037
	523	39.73	799	19.95	1716	0.026
	573	85.81	712	46.15	1405	0.014

\*\*Data collected from Ref. [15]



Nickel			Ceramic		
Electrolytic	37-260	0.04-0.06	Earthenware, glazed	20	0.90
Pure, polished	260	0.07	Earthenware, matte	20	0.93
Oxidized at 600°C	260-540	0.37-0.48	Porcelain	22	0.92
Platinum			Refractory, black	93	0.94
Plate, polished	260-540	0.06-0.10	Clay, fired	70	0.91
Filament	26-1225	0.04-0.19	Concrete, rough	37	0.94
Silver			Glass		
Polished	37-625	0.02-0.03	Smooth	22	0.94
Stainless steel			Pyrex, lead, and soda	260-530	0.95-0.85
Polished	23	0.17	Ice		
Cleaned	23	0.21-0.39	Smooth	0	0.97
Tin			Rough crystals	0	0.985
Polished	37	0.05	Marble, light gray, polished	22	0.93
Tungsten			Mica	37	0.75
Filament	3300	0.39	Paints		
Zinc			Aluminum 10%, Lacquer 22%	100	0.52
Polished	225-325	0.05-0.06	Aluminum 26%, Lacquer 27%	100	0.30
Oxidized at 400°C	400	0.11	Other aluminum Paints	100	0.27
Galvanized	23-27	0.23-0.28	Lacquer, white	100	0.925
<i>Non-metals</i>			Lacquer, black matte	80	0.97
Alumina (85-99.5% Al <sub>2</sub> O <sub>3</sub> ), effect of mean grain size			Oil paints, all colors	100	0.92-0.96
10 μm	1000-1560	0.30-0.18	Oil paints	20	0.89-0.97
50 μm	1000-1560	0.39-0.28	Paper		
100 μm	1000-1560	0.50-0.40	Ordinary	20-95	0.80-0.92
Asbestos			Tar	20	0.93
Paper	37	0.93	Porcelain, glazed	22	0.92
Board	37	0.96	Quartz		
Brick			Glass, 1.98 mm thick	280	0.90
Magnesite, refractory	1000	0.38	Glass, 1.98 mm thick	840	0.41
Red, rough	21	0.93	Glass, 6.88 mm thick	280	0.93
Gray, glazed	1100	0.75	Glass, 6.88 mm thick	840	0.47
Silica	540	0.80	Rubber		
Carbon			Hard	23	0.94
Filament	1050-1400	0.526	Soft, gray	23	0.86
Candle soot	95-270	0.952	Soil	37	0.93-0.96
Lampblack	20	0.93-0.967	Water, deep	0.-100	0.96
			Wood	20	0.80-0.90

Constructed based on the data compiled in Refs. [16, 23]

**Table 2.19** Solar absorptivity of surfaces (receiving surface at room temperature)\*\*

Surface	$\alpha$	Surface	$\alpha$
<b>Metals</b>		<b>Non metals</b>	
Aluminum		Asphalt	
Polished	0.10	Pavement	0.85
Anodized	0.14	Pavement free from dust	0.93
Foil	0.15	Brick	
Brass		White glazed	0.26
Polished	0.3-0.5	Red	0.70-0.77
Dull	0.4-0.65	Concrete	
Chromium, electroplated	0.41	Uncolored	0.65
Copper		Brown	0.85
Highly polished	0.18	Black	0.91
Clean	0.25	Earth, plowed field	0.75
Tarnished by exposure	0.64	Granite	0.45
Gold	0.21	Grass	0.75-0.8
Iron		Gravel	0.29
Matte, oxidized	0.96	Leaves, green	0.71-0.79
Lead roofing, old	0.77	Magnesium oxide (MgO)	0.15
Nickel		Marble	
Highly polished	0.15	White	0.44
Polished	0.36	Ground, unpolished	0.47
Oxidized	0.79	Paints	
Platinum, bright	0.31	Oil, white lead	0.24-0.26
Silver		Oil, light cream	0.30
Highly polished	0.07	Oil, light green	0.50
Polished	0.13	Aluminum	0.55
Stainless steel, type 301		Oil, light gray	0.75
Polished	0.37	Oil, black on galvanized iron	0.90
Clean	0.52	Paper	
Zinc		White	0.28
Highly polished	0.34	Sand	0.76
Polished	0.55	Sawdust	0.75
		Slate	
		Silver gray	0.79
		Blue gray	0.85
		Dark gray	0.90
		Snow, clean	0.2-0.35
		Soot, coal	0.95
		Zinc oxide	0.15

\*\*Data collected from Refs. [23]

Table 2.20 Hyperbolic Functions

$x$	$\sinh x$	$\cosh x$	$\tanh x$	$x$	$\sinh x$	$\cosh x$	$\tanh x$
0	0	1	0	2	3.6269	3.7622	0.96403
0.1	0.1002	1.005	0.09967	2.1	4.0219	4.1443	0.97045
0.2	0.2013	1.0201	0.19738	2.2	4.4571	4.5679	0.97574
0.3	0.3045	1.0453	0.29131	2.3	4.937	5.0372	0.9801
0.4	0.4108	1.0811	0.37995	2.4	5.4662	5.5569	0.98367
0.5	0.5211	1.1276	0.46212	2.5	6.0502	6.1323	0.98661
0.6	0.6367	1.1855	0.53705	2.6	6.6947	6.769	0.98903
0.7	0.7586	1.2552	0.60437	2.7	7.4063	7.4735	0.99101
0.8	0.8881	1.3374	0.66404	2.8	8.1919	8.2527	0.99263
0.9	1.0265	1.4331	0.7163	2.9	9.0596	9.1146	0.99396
1	1.1752	1.5431	0.76159	3	10.018	10.068	0.99505
1.1	1.3356	1.6685	0.8005	3.5	16.543	16.573	0.99818
1.2	1.5095	1.8107	0.83365	4	27.29	27.308	0.99933
1.3	1.6984	1.9709	0.86172	4.5	45.003	45.014	0.99975
1.4	1.9043	2.1509	0.88535	5	74.203	74.21	0.99991
1.5	2.1293	2.3524	0.90515	6	201.71	201.72	0.99999
1.6	2.3756	2.5775	0.92167	7	548.32	548.32	1
1.7	2.6456	2.8283	0.93541	8	1490.5	1490.5	1
1.8	2.9422	3.1075	0.94681	9	4051.5	4051.5	1
1.9	3.2682	3.4177	0.95624	10	1101.3	1101.3	1

The hyperbolic functions are defined as  
 $\sinh x = \frac{1}{2}(e^x - e^{-x})$ ;  $\cosh x = \frac{1}{2}(e^x + e^{-x})$ ;  $\tanh x = \frac{\sinh x}{\cosh x}$

**Table 2.21** Bessel functions of the first kind and second kind

<i>I<sub>0</sub>(y)</i> Zero order Bessel function of first kind										
<i>y</i>	0	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9
0	1	1.0025	1.01	1.0226	1.0404	1.0635	1.092	1.1263	1.1665	1.213
1	1.2661	1.3262	1.3937	1.4693	1.5534	1.6467	1.75	1.864	1.9896	2.1277
2	2.2796	2.4463	2.6291	2.8296	3.0493	3.2898	3.5533	3.8417	4.1573	4.5027
3	4.8808	5.2945	5.7472	6.2426	6.7848	7.3782	8.0277	8.7386	9.5169	10.369
4	1.1302	1.2324	1.3442	1.4668	1.601	1.7481	1.9093	2.0858	2.2794	2.4915
5	2.724	2.9789	3.2584	3.5648	3.9009	4.2695	4.6738	5.1173	5.6038	6.1377
6	6.7234	7.3663	8.0718	8.8462	9.6962	10.629	11.654	12.779	14.014	15.37
10 <sup>2</sup> ×	1.6859	1.8495	2.0292	2.2266	2.4434	2.6816	2.9433	3.2309	3.5468	3.8941
10 <sup>3</sup> ×	4.2756	4.695	5.1559	5.6626	6.2194	6.8316	7.5046	8.2445	9.058	9.9524
10 <sup>4</sup> ×	1.0936	1.2019	1.3207	1.4514	1.5953	1.7535	1.9275	2.1189	2.3294	2.5611

$$I_0(y) \approx \frac{0.3989 e^{y^2}}{y^2} \left\{ 1 + \frac{1}{8y} + \frac{9}{128y^2} + \frac{75}{1024y^3} \right\}$$

<i>K<sub>0</sub>(y)</i> Zero order Bessel function of second kind										
<i>y</i>	0	0.1	0.2	0.3	0.4	0.5	0.6	0.7	0.8	0.9
0	∞	2.4271	1.7527	1.3725	1.1145	0.9244	0.7775	0.6605	0.5653	0.4867
1	0.421	0.3656	0.3185	0.2782	0.2437	0.2138	0.188	0.1655	0.1459	0.1288
10 <sup>-1</sup> ×	1.1389	1.0078	0.8926	0.7914	0.7022	0.6235	0.554	0.4926	0.4382	0.3901
10 <sup>-1</sup> ×	0.3474	0.3095	0.2759	0.2461	0.2196	0.196	0.175	0.1563	0.1397	0.1248
10 <sup>-2</sup> ×	1.116	0.998	0.8927	0.7988	0.7149	0.64	0.573	0.5132	0.4597	0.4119
10 <sup>-2</sup> ×	0.3691	0.3308	0.2966	0.2659	0.2385	0.2139	0.1918	0.1721	0.1544	0.1386
10 <sup>-3</sup> ×	1.244	1.1167	1.0025	0.9001	0.8083	0.7259	0.652	0.5857	0.5262	0.4728
10 <sup>-3</sup> ×	0.4248	0.3817	0.3431	0.3084	0.2772	0.2492	0.224	0.2014	0.1811	0.1629
10 <sup>-4</sup> ×	1.4647	1.3173	1.1849	1.0658	0.9588	0.8626	0.7761	0.6983	0.6283	0.5654
10 <sup>-4</sup> ×	0.5088	0.4579	0.4121	0.371	0.3339	0.3006	0.2706	0.2436	0.2193	0.1977

$$K_0(y) \approx \frac{1.2533 e^{-y}}{y^2} \left\{ 1 - \frac{1}{8y} + \frac{9}{128y^2} - \frac{75}{1024y^3} \right\}$$

**Table 2.22** Gaussian Error Function

$x/2\sqrt{\alpha\tau}$	$\text{erf}(x/2\sqrt{\alpha\tau})$	$x/2\sqrt{\alpha\tau}$	$\text{erf}(x/2\sqrt{\alpha\tau})$	$x/2\sqrt{\alpha\tau}$	$\text{erf}(x/2\sqrt{\alpha\tau})$
0.00	0.00000	0.76	0.71754	1.52	0.96841
0.02	0.02256	0.78	0.73001	1.54	0.97059
0.04	0.04511	0.80	0.74210	1.56	0.97263
0.06	0.06762	0.82	0.75381	1.58	0.97455
0.08	0.09008	0.84	0.76514	1.60	0.97636
0.10	0.11246	0.86	0.77610	1.62	0.97804
0.12	0.13476	0.88	0.78669	1.64	0.97962
0.14	0.15695	0.90	0.79691	1.66	0.98110
0.16	0.17901	0.92	0.80677	1.68	0.98249
0.18	0.20094	0.94	0.81627	1.70	0.98379
0.20	0.22270	0.96	0.82542	1.72	0.98500
0.22	0.24430	0.98	0.83423	1.74	0.98613
0.24	0.26570	1.00	0.84270	1.76	0.98719
0.26	0.28690	1.02	0.85084	1.78	0.98817
0.28	0.30788	1.04	0.85865	1.80	0.98909
0.30	0.32863	1.06	0.86614	1.82	0.98994
0.32	0.34913	1.08	0.87333	1.84	0.99074
0.34	0.36936	1.10	0.88020	1.86	0.99147
0.36	0.38933	1.12	0.88079	1.88	0.99216
0.38	0.40901	1.14	0.89308	1.90	0.99279
0.40	0.42839	1.16	0.89910	1.92	0.99338
0.42	0.44749	1.18	0.90484	1.94	0.99392
0.44	0.46622	1.20	0.91031	1.96	0.99443
0.46	0.48466	1.22	0.91553	1.98	0.99489
0.48	0.50275	1.24	0.92050	2.00	0.995322
0.50	0.52050	1.26	0.92524	2.10	0.997020
0.52	0.53790	1.28	0.92973	2.20	0.998137
0.54	0.55494	1.30	0.93401	2.30	0.998857
0.56	0.57162	1.32	0.93806	2.40	0.999311
0.58	0.58792	1.34	0.94191	2.50	0.999593
0.60	0.60386	1.36	0.94556	2.60	0.999764
0.62	0.61941	1.38	0.94902	2.70	0.999866
0.64	0.63459	1.40	0.95228	2.80	0.999925
0.66	0.64938	1.42	0.95538	2.90	0.999959
0.68	0.66278	1.44	0.95830	3.00	0.999978
0.70	0.67780	1.46	0.96105	3.20	0.999994
0.72	0.69143	1.48	0.96365	3.40	0.999998
0.74	0.70468	1.50	0.96610	3.60	1.000000

The Gaussian error function is defined as

$$\text{erf}(x) = \frac{2}{\sqrt{\pi}} \int_0^x e^{-v^2} dv$$

$$\text{erf}(\infty) = 1$$